



"ANALYSIS ON RAILWAY SLEEPERS MANUFACTURED FROM POLYMERS AND IRON SLAG"

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ABSTRACT

Timber is the most commonly used material for railway sleepers. However, as a sleeper material deteriorates with time, it becomes so expensive and needs suitable replacement. There are also now various environmental concerns regarding the use and disposal of chemically-impregnated timber sleepers. Composite sleepers have become great replacement of existing timber ones. In this paper, a review of advantages and disadvantages of existing timber sleepers are presented as well as conventional methods in analysis and design of railway sleepers. Trials and proportions of proposed composite mixture using polymers and iron slag are portrayed. Mechanical and physical testing has been carried out to determine the important properties of the new selected mixture; hence results are compared with the recommendations of the international railway standards. It is found that the testing results are larger than the recommended values. Thus; it is concluded that the tested composite mixture is effective and adequate to be used in manufacture of railway sleepers. Selection of the optimal sleeper dimensions by Evaluation of 21 suggested models of sleeper with different dimensions is also presented.

KEYWORDS: Railway Sleepers, fiber composites, Iron Slag, mechanical testing.

1. INTRODUCTION

Railway sleeper is one of the most important components of the railway system. It is a beam under the rails as shown in Fig.1[1] to support the track and keep the required gauge width. It is also responsible for distribution and transfer of load to ballast section, and prevent any lateral and longitudinal movement of rail system [2].

Timber sleepers which are used in Egyptian National Railways (ENR) are manufactured from one of three different types; Beach, Azobe, or Oak woods [3]. The main advantage of timber sleepers is their adaptability. They can be fitted with all types of railway track. Timber sleepers are workable, easy to handle, easy to replace and needs no complex assembly equipment. Thus, local problem sites can be repaired or replaced without the need for outside support which is represented in either manpower or equipment. It is only appropriate for low speed lines with the speed limit 160 km/h [4]. In addition, it can absorb severe impact with limited damage [5].

Timber sleeper has an excellent electrical isolation, an important factor for track signaling which cannot be matched by other alternative sleeper materials except maybe by plastics or fiber composites[6].

The main disadvantage in using timber sleepers is its high cost and their exposure to mechanical and biological degradation leading to failure [7]. Fungal decay, end splitting and spike retention are most common failure modes in timber sleepers as shown in Figures 2.a [8]& 2.b [7]. Timber sleepers were also attacked and damaged by termites [7]. Cracking at fasteners positions can happen during handling or installation as shown in Fig. 2.c [9]. An alternative material for sleeper replacement to reduce maintenance cost and overcome problems encountered using timber sleepers is therefore both desirable and necessary.

Fiber composites could be an ideal material for the development of railway sleepers. This composite material typically comprises of strong fibers embedded within a light polymer matrix offering high strength, lightweight, durability, good electrical insulating properties and low-life maintenance costs which is a suitable material for the replacement of deteriorated timber sleepers [10].

The performance of a sleeper to resist lateral and longitudinal loading is relied on the sleeper's size, shape, surface geometry, weight, and spacing [11]. Current practices concerning the analysis and design of sleepers include three steps as following [12]: a) considering a dynamic coefficient; b) estimation of vertical rail seat load; c) assuming a stress distribution pattern under the sleeper and applying vertical static equilibrium to a structural model of the sleeper.

Researchers all over the world have recommended several formulae and values for the calculation of the dynamic coefficient. Table 1 presents a summary of the main recommendations for the dynamic coefficient factors [12]. The exact magnitude of the load applied to each rail seat depends upon numerous parameters including the rail weight, the sleeper spacing, the track modulus per rail, the amount of play between the rail and sleeper, and the amount of play between the sleeper and ballast [11]. Based on these considerations, various relations are proposed for the amount of rail-seat loads that are summarized in Table 2.

The general approach for the calculation of the contact pressure beneath the rail seat is to assume a uniform contact, pressure distribution over the assumed effective area of the sleeper. This assumption is made in order to facilitate the ease of calculations. While maximum allowable pressure between the sleeper and the ballast can be calculated according to AREA formula [13] as follow:

$$P_a = \frac{4 P D F}{B l}$$

$P_a =$

Where P = design wheel load (kN) (i.e. static wheel load multiplied by the dynamic coefficient), DF = the AREA distribution factor, B = breadth of sleeper (m), and l = total length of sleeper (m).

Bending stress at rail seat is calculated according to following equation [14]:

$$\sigma_{ru} = \frac{M_r}{Z} = 3 * q_r *$$

Where: Z (section modulus) = $\frac{Bt^2}{6}$, q_r = rail seat load (KN), l = total sleeper length (m), g = distance between rail centers (m), B = sleeper width (m), t = sleeper thickness (m), M_r = bending moment at rail seat which is calculated with different methods as shown in Table 3.

Bending stress at center of sleeper is calculated according to the following equation [14] by assuming uniform pressure distribution over the total sleeper length.

$$\sigma_c = \frac{M_c}{Z} = \frac{3}{2} * q_r *$$

M_c = bending moment at center of sleeper which is calculated with different methods as shown in Table 3.

2. Experimental program

2.1 Materials

Recycled high density polyethylene (RHDPE), iron (blast furnace) slag [15], calcium carbonate (CACO₃) and polyester resin are used in addition to E glass fiber with different type and weight for getting a composite mixture which complies with universal standards.

2.2 Mixture proportions

The mixture proportions of the tested composite mixtures which are divided into four groups are given in Table 4.

2.3 Mixing Stage

A hand lay-up (HLU) technique was used to mix and cast the specimens. The solids constituents of the composite mixture, the recycled high density polyethylene, steel slag and calcium carbonate were dry mixed for about three minutes. The liquid part of the mixtures, the polyester resin, the styrene and cobalt were premixed then added to the solids. The wet mixing usually continued for another four minutes to achieve a uniform dispersion of mixture components. Before casting the mixture into the mold, MEKP was added and thoroughly stirred.

2.4 Casting and Curing Stage

The fiber-plastic molds were manufactured at private workshop. The molds were in the form of 100x150x150 mm cubes and 100 x100 x500 mm prisms as shown in Fig. 5 [15]. Once the mixture had been mixed, the mixture was poured into the molds in three or four layers according to number of used fiber layers as shown in Fig. 6. Three cubes were taken from each mix (100x150x150 mm) and 2 flexure beams (100x100x500mm).

After casting stage, the specimens were stored in the laboratory. Then, after 24 hours, the specimens were demolded.

3. Testing procedure

Mechanical and physical tests were performed on composite specimens in accordance with the American Society for Testing and Materials (ASTM) standards.

3.1 Compressive strength

Compressive test perpendicular and parallel to grain of the specimens was performed following the (ASTM D 6108-97) standards [16] at Properties of Materials and Quality Control Lab, Shoubra Faculty of Engineering. 100 x 150 x150 mm cubical specimens were used for determination of compressive strength and Stress- strain relationship for composite mixes. The load was applied uniformly on the loading surface of the specimen at a constant cross head speed of 1mm/min till the failure. The test set-up for compressive test of the composite specimen is shown in Fig.7.

3.2 Flexural strength (modulus of rupture)

Flexural strength test was carried out on prismatic specimens of 100 x 100 x 500 mm. The 4-point static bending test on composite specimens was performed in accordance with the (ASTM D6109 -97) standards at Properties of Materials and Quality Control Lab, Shoubra Faculty of Engineering by the use of a 50 KN manual hydraulic flexural machine as shown in Fig.8. The load was applied at the third and at the two-third points of the span of specimen with rate 3mm/ min. Schematic illustration of test set up is shown in Fig.9. The flexural strength in terms of modulus of rupture was calculated using the following equation:

$$R = \frac{PL}{bd^2} \text{ (ASTM D6109 -97)[16]}$$

Where; R= Modulus of rupture (N/mm²), P = Maximum (failure) load (N), L = effective Length of the specimen (mm), b = width of the specimen (mm), d = depth of the specimen (mm).

3.3 Load - Deflection relationship

To get Load -deflection relationship, the test was performed on mix S16 accordance with (ASTM D6109 -97) standards [16] at Researches and Building Center. The 4 - point static bending test was performed on composite beams as illustrated in Fig.10. The load was applied at the third and at the two-third points of the span of specimen with rate 3mm/ min. Stress and strain are computed by using the following formulae according to ASTM D6109 -97 standards to get modulus of elasticity flexure

$$\text{Stress} = (P_2 - P_1) * L / bd^2$$

$$\text{Strain} (\epsilon) = 4.7 * (\Delta_2 - \Delta_1) / L^2$$

$$\text{Modulus of elasticity (E)} = \text{stress} / \text{strain}$$

Where:

P = Maximum (failure) load (N), L = effective Length of the specimen (mm), b = width of the specimen (mm), d = depth of the specimen (mm), Δ = deflection (mm)

3.4 Density and Specific Gravity Test

This test was performed on four samples of mix S16 according to (ASTM D 6111-03) [16] at Railway Engineering Lab, Shoubra Faculty of Engineering. The four samples are illustrated in Fig.11. The specimen is weighed in air then weighed when immersed in distilled water at 23°C using a sinker and wire to hold the specimen completely submerged as required. Density and Specific Gravity are calculated according to following equation:

- **Specific gravity** = $a / [(a + w) - b]$ (ASTM D 6111-03)
a = mass of specimen in air, b = mass of specimen and sinker (if used) in water.
W = mass of totally immersed sinker if used and partially immersed wire
- **Density (kg/m³)** = (specific gravity) $\times \rho_{\text{water}}$
Where: $\rho_{\text{water}} = 997.6 \text{ Kg/m}^3$
- **Density (kg/m³)** = (specific gravity) $\times (997.6)$

3.5 Coefficient of linear thermal expansion (CLTE) Test

The test was carried out at Chemistry Administration on Ramses according to (ASTM D 696-03) [16] which covers temperatures between -30°C and 30°C. 25 *25* 150 mm specimen was used as shown in Fig.12. The Specimen is placed in a -30°C (-22°F) constant temperature bath. After the specimen has reached a temperature of -30°C, the constant temperature bath is replaced by a 30°C (86°F) constant temperature bath. After the specimen has reached a temperature of 30°C, the 30°C bath is replaced by the -30°C bath. After the specimen has reached a temperature of -30°C, the specimen is removed and measured at room temperature. Test set up is shown in Fig. 14.

CLTE (α) is calculated using the formula:

$$\alpha = \Delta L / (L_0 * \Delta T) \text{ (ASTM D 696-03)}$$

Where: ΔL is the change in length of the specimen, L_0 is the original length of the specimen and ΔT is the temperature change during the test.

4. Results and Discussions

The methodology of composite mix selection is based on economy and achieving both compressive and flexural strength that satisfies the recommendations of American Railway Engineering and Maintenance of way Association (AREMA) standards and Chicago Transit Authority (CTA) specifications [17]. AREMA standards and CTA specifications are presented in Table 5.

4.1 Compression test perpendicular to grain

The reported values for both density and compressive strength of composite mixtures are presented in Table 6.

The results in Table (6) show that the compressive strength ranged from 23.70 MPa to 32.08 MPa. It is also concluded that all mixes have a value more than Chicago (CTA) specifications and AREMA standards which recommend that minimum compressive strength is 6.89 , 6.2 MPa. The optimum mix from all mixes is mix S16 that has a value of compressive strength much higher than recommended minimum values of Chicago (CTA) and AREMA standards. S16 has also lower cost of used materials which make it more economy than other mixes. The specimen tested under perpendicular compression to grain failed by formation of cracks appeared at the center of its side as illustrated in Fig.14.

4.2 Stress- Strain relationship

Figures from (15) to (24) illustrate the stress-strain curves of mixes containing different percentages of matrix, fillers and additives. The stress- strain curve and modulus of elasticity was measured on mixes S1, S2, S4, S4w, S5, S1-c, S8, S10, S4c and S16. The elasticity modulus was calculated from the slope of the initial portion of the compressive stress-strain relation curve. Compressive modulus of elasticity is shown in Table 7.

The stress-strain curve in Figures (15 to 24) shows that the composite mixtures behave almost linearly under compression at higher level of stress. With the continuous application of load, the composite material started to behave nonlinearly up to failure. A sudden drop in the stress level was observed at a strain between 4500 and 7000 micro strains. The sudden drop in the stress level could be due to the cracks which develop at the outer part of the composite sample. Shortly after the maximum compressive stress is reached, the crack width increased and the stress level dropped dramatically. The specimen failed with all the cracks occurring at the outer part of the sample. The failure strain was not measured for mix S4w. It is due to cracks that happened in the outer wrapped fiber which led to failure of attached strain gauge before specimen failure.

From stress- strain results, it is clear that the values of modulus of elasticity (E_c) which were calculated from the stress-strain curve by chord method ranged from 3427 MPa to 10297 MPa. The lower and higher values of modulus of elasticity (E_c) specified to the mix S5 and mix S2 respectively. It is concluded that all mixes have a value more than AREMA standards and Chicago (CTA) specifications which recommend that minimum modulus of elasticity is 1172.11 MPa. Mix S16 is preferred to be the proposed composite mixture. Because it combines with low materials cost and high value of elasticity modulus that is 8581 MPa which is higher than a minimum recommended value of AREMA and Chicago (CTA) standards.

The results also indicated that the relationship of the stress-strain and strain values of the proposed mix S16 which are 5000 and 6000 micro strain are similar to the behavior of some wooden types.

4.3 Flexural strength

The reported values for flexural strength of composite mixtures are presented in Table 8. Flexural load and strength were obtained for two specimens after the failure.

The flexure strength can be expressed in terms of modulus of rupture, where the maximum stress is at rupture. The results of flexure strength in Table 8 ranged from 22.07 MPa to 10.95 MPa. The higher value of flexure strength was recorded for the mix S16 and the lowest value of flexure strength was achieved by mix S4. Mix S16 was selected to be the optimum mixture. Because it has a highest value of flexure strength which is higher than minimum recommended values of AREMA standards and Chicago (CTA) specifications which are 13.7 MPa and 17.23 MPa respectively.

4.4 Load – deflection relationship

The reported values for maximum flexural load and modulus of elasticity of composite mixtures are presented in Table 13. Also, load – deflection curve is illustrated for mixes S16-1, S16-2 in Fig. 25.

From Figure 25 , it is noted that the load of specimen S16-1 and S16-2 increased linearly with deflection until a load of 22 KN where a slight decrease in stiffness was observed after. It is due to the initiation of tensile cracks in the core of specimen at the constant moment region. Flexural cracks were observed at the bottom of specimen as shown in Fig.26. The specimen S16-1 failed at an applied load of around 56 KN with a mid-span deflection 6 mm. but specimen S16-2 failed at an applied load of around 54 KN with a mid-span deflection 6 mm. It can also be observed that the specimen S16 has an average flexural elasticity modulus of around 1519.82 MPa which is larger than recommended values of Chicago (CTA) and AREMA standards that are 1379 ,1172 MPa respectively.

4.5 Compressive strength parallel to grain

The reported values for maximum compressive load parallel to grain of composite mixture S16 are presented in Table 9. Compressive load and strength were obtained for three specimens of S16 after the failure.

From table 9, it is clear that the specimen has a compressive strength value that is higher than 20.68 MPa which is a recommended minimum value of Chicago standards.

4.6 Density and specific gravity

Density and specific gravity results for four samples of mix S16 were obtained as shown in Table 10.

From specific gravity test results, it is concluded that density of proposed sleeper which is manufactured from mix S16 is 1314 Kg/m³. The weight of proposed sleeper meets or falls within the required weight – range specified for composite sleepers in AREMA-30 standard [18].

4.7 Coefficient of linear thermal expansion

First trial, after the specimen has subjected to cold (-30°C) – hot (30) – cold (-30) series of exposure conditions for 24 hours, no change in length was found.

Second trial , After the specimen has placed in -30°C path for 30 hours then in 30 C for 30 hours then -30 C for 30 hours, slightly change in length was occurred as shown in table 16

It is obviously that the specimen has a low coefficient of linear thermal expansion 2.3E-05 mm/mm.C^o which less than recommended maximum values of AREMA and Chicago standards that is 13.5E-05 and 12.6 E-05 respectively.

5. Design of proposed sleeper

Sleeper with 170 mm thickness, 230mm width and 2600 mm length (T1) was used as a datum to reach to the optimum model which satisfies standards recommendations. In order to reach the optimum geometrical properties of a T1 sleeper, 21 suggested dimensions of T1 are evaluated as shown in Table 5-1.

For the analysis of the models, the following assumptions are taken into account:

- The maximum transferred load to the sleeper exactly under the wheel is 50% of the wheel load according to (three adjacent sleeper)[12] method.
- The design speed of trains is 120 Km/hr.
- The dynamic factor \emptyset which was calculated according to Schramm& DB equation [12] is 1.388.
- The axial load is considered to be 22.5 Tons.
- The concentrated load at each rail seat position according to (Three Adjacent Sleeper) method is 76.51 KN: $q_r = 0.5 \times 1.388 \times 11250 \times 9.8 = 76.51 \text{ kN}$.
- The maximum contact pressure between sleeper and ballast was calculated from AREA

$$\frac{A P \emptyset D . F}{B l}$$

formula [13]: $P_a =$

- Positive Bending moment at rail seat was calculated according to Australian equation [14]:

$$M_r = q_r \left(\frac{l - g}{8} \right).$$

- Negative bending moment at center of sleeper was calculated according to Raymond's

equation [14]: $M_c = q_r \left(\frac{2g - l}{4} \right)$

- Bending stress at rail seat and center of sleeper were calculated according to formula σ_{ru} or σ_c = [14]

- Minimum thickness at rail seat was calculated according to AREA method [11]:

$$T_{\min}^2 = \frac{3l * (l - g) * P_{all}}{8 * 1000 * \sigma_{ru} * F_2^2}$$

where P_{all} = maximum allowable pressure between ballast and sleeper which is ranged from 450 KPa to 590 KPa according to AREA recommendations [13], F_2 is safety factor depending on track maintenance which equal 2 according to AREA recommendations.

- Minimum thickness at center of sleeper was calculated from AREA equation [11]: $T_{\min}^2 =$

$$\frac{3l * (2g - l) * P_{all}}{4 * 1000 * \sigma_c * F_2^2}$$

5.1 Results analysis

Bending stresses and sleeper-ballast contact pressure were considered as the design criterion. These parameters were determined for the proposed models as shown in Table 13 using previous equations to select the preliminary optimal model of sleeper. The results are then compared and checked with their allowable limits as discussed in the following sections.

5.1.1 Bending stress at rail seat

From results, it is noted that all proposed models have bending strength at rail seat greater than minimum AREMA recommendation (6.8 MPa) as shown in Figure 26.

5.1.2 Bending stress at center of sleeper

From results, it is obvious that the models T9 and T18 have a bending stress greater than minimum AREMA recommendations (13.8 MPa) as shown in Figure 27.

5.1.3 Contact pressure between sleeper and ballast

From results, it is obvious that all the models have a contact pressure within the limits of maximum allowable pressure according to AREA recommendations (590KPa) except model T21 as shown in Figure 28.

5.2 Selection of the optimal sleeper dimensions

Selecting the optimum model is based on the criteria mentioned above, the bending stress and the pressure distribution beneath the sleeper. It is found that T9 and T18 which have geometrical properties with 150 mm depth at center of sleeper and 2400 mm length is the optimal design. T9 with 230mm width is preferred because it has contact pressure between ballast and sleeper lesser than T18. Then, It was decided that the sleeper design with rail seat inclination 1:20 was selected, so the depth of sleeper was designed to be changed from 170 at the end of sleeper to 150 at center of sleeper with average depth 160 mm at rail seat position as shown in Figure 29 . This proposed design of sleeper was decided to be manufactured to perform recommended laboratory tests on it. Actual manufactured sleeper with proposed design is portrayed in Figure 30.

6. CONCLUSIONS

In terms of strength and stiffness according to the comparison of test results with the standard values AREMA and CTA, the material of the proposed innovated sleeper is adequate to sustain the loading conditions of Egyptian National Railways (ENR). The main findings of proposed composite mixture based on testing results can be summarized as follows:

- 1- The relationship of the stress-strain and values of strain of the proposed material are similar to the behavior of some wooden types that used in manufacturing of railway sleepers.

- 2- The proposed composite mixture illustrated linear elastic up to the proportion limit in the compression test.
- 3- Obtained value of maximum compression stress (perpendicular to grain), 32.08MPa, is larger than the required value by the standards of railways.
- 4- Obtained value of maximum compression stress (parallel to grain), 36.39 MPa, is larger than the required value by the standards of railways.
- 5- Obtained value of modulus of elasticity (compression), 8581 MPa, is larger than the required value by the standards of railways.
- 6- Obtained value of flexure strength, 22.07 MPa is larger than the required value by the standards of railways.
- 7- Obtained value for the modulus of elasticity (flexural) is 1519 MPa.
- 8- The proposed composite mixture has a coefficient of linear thermal expansion, $2.3E-05$ mm/mm/C° which is less than recommended values of standards of railways.
- 9- Selection of the optimal model of sleeper dimensions to be manufactured based on evaluation of 21 proposed models with different dimensions.

However; samples of full scale sleepers should be tested in static and dynamic conditions; hence the results should be compared with AREMA and ENR standards.

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Table 1 : Recommended relationship for dynamic coefficient factors [12]

Recommender	Relation
AREA	$\phi = 1 + 5.21 \frac{V}{D}$
Eisenmann	$\phi = 1 + \delta \eta t$
ORE	$\phi = 1 + \alpha' + \beta' + \gamma'$
DB	$\phi = 1 + \frac{V^2}{30000}$, $\phi = 1 + \frac{4.5V^2}{10^3} - \frac{1.5V^3}{10^7}$
BR	$\phi = \frac{8.784(\alpha_1 + \alpha_2) V}{P_s} \left[\frac{D_j P_u}{g} \right]^{1.2}$
India	$\phi = 1 + \frac{V}{58.14k^{0.5}}$
South Africa	$\phi = 1 + 4.92 \frac{V}{D}$
CA	$\phi = 1 + \frac{19.65V}{D K \sqrt{2}}$
WMMTA	$\phi = (1 + 3.86 * 10^{-5} V^2)^{0.67}$
SADEGHI	$\phi = 1.098 + 8 \times 10^{-4} V + 10^{-6} V^2$

Table 2 : Relations for the calculation of maximum rail seat load [12].










Maximum rail seat load (KN)	Methods
$q_r = 0.5p$	Three adjacent sleepers method
$q_r = 0.43p$	Australian Formula [ARS]
$q_r = 0.6p$	AREA method
$q_r = 0.65p$	ORE method

Table 3: Relations for the bending moment calculation [13].

Center moment		Rail seat moment		Developer	Sleeper Type
M_r^- (KN.m)	M_r^+ (KN.m)	M_r^- (KN.m)	M_r^+ (KN.m)		
$qr \left(\frac{g}{2} \right)$	-----	-----	$qr \left(\frac{l-g}{2} \right)$	Battelle	Timber
-----	-----	-----	$qr \left(\frac{l-g-j}{8} \right)$	Scharmm	
$qr \left(\frac{2g-l}{4} \right)$	-----	-----	-----	Raymond	
$qr \left(\frac{2g-l}{4} \right)$	$0.05 \times qr \times (-g)$	-----	$qr \left(\frac{l-g}{8} \right)$	Australian standard	Steel
$qr \left(\frac{2g-l}{4} \right)$	$0.05 \times qr \times (-g)$	Max{ 0.67 $M_r^+, 14$ }	$qr \left(\frac{l-g}{8} \right)$	Australian standard	Concrete

Table 4 : Trials of composite mixture

(Note: the following materials percentages represent percentages of total mixture weight expect cobalt and MEKP)

Fiber distribution	Fiber distribution illustration	Styrene %	Polyester resin %	CaCo3 %	Steel slag %	RHDPE %	Mix	Group
Equally along depth of specimen		---	58	---	21	21	S1	Group 1
		---	52	---	24	24	S2	
Equally along depth of specimen		9	43	---	24	24	S4	Group 2
Equally along depth of specimen		9	43	---	24	24	S4w	
Equally along depth of specimen		9	32	---	35	24	S5	
Equally along depth of specimen		15	26	---	35	24	S6	
Equally along depth of specimen		---	50	10	20	20	S1c	Group 3
2 compacted layers at bottom and one at upper		---	48	14	18	20	S8	
3 compacted layers at bottom and one at upper		---	48	14	18	20	S7	
2 compacted layers at bottom and one at upper		---	44	18	18	20	S10	

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3 compacted layers at bottom and one layer at upper		—	44	18	18	20	S11	Group 4
Equally along depth of specimen		8	38	10	22	22	S4c	
3 compacted layers at bottom and one upper		5	35	20	20	20	S14	
3 laminates at bottom and one upper		5	35	20	20	20	S16	

Table 5 : AREMA and CTA composite sleepers' standards [17]

AREMA Standards	Chicago Transit Authority (CTA) specifications	Mechanical Properties/ Test Method
0.000135 mm/ mm. C ⁰ (Max)	0.000126 mm/mm. C ⁰ (Max)	Coefficient of Thermal Expansion (ASTM D696-98)
—	20.68 MPa (Min.)	Compressive Strength ASTM D6108-97 (Compression Parallel to Grain)
6.2 MPa (Min.)	6.8 MPa (Min.)	Rail Seat Compression Perpendicular to Grain ASTM D6108-97
1172 MPa (Min.)	1172 MPa (Min.)	Modulus of Elasticity (Compression) ASTM D6108-97
13.8 MPa (Min.)	17.23 MPa (Min.)	Flexural Strength ASTM D6109-97
1172 MPa (Min.)	1379 MPa (Min.)	Modulus of Elasticity (Flexural) ASTM D6109-97

Table 6 : Compressive strength results of composite mixtures

Average compressive strength (MPa)	Compressive strength (MPa)	Compressive load (KN)	Density Kg/m ³		Group
31.91	33.73	506	1210	S1-1	Group 1
	29.00	435		S1-2	
	33.00	495		S1-3	
27.11	27.53	413	1250	s2-1	Group 2
	27.00	405		S2-2	
	26.80	402		S2-3	
26.48	27.40	411	1230	S4-1	Group 2
	25.33	380		S4-2	
	26.73	401		S4-3	
25.71	26.40	396	1250	S4w-1	Group 2
	25.33	380		Sw-2	
	25.40	381		Sw-3	
23.70	24.66	370	1640	S5-1	Group 2
	21.33	320		S5-2	






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	25.13	377		S5-3	
	failed			S6	
28.28	30.20	453	1230	S1c-1	Group 3
	27.20	408		S1c-2	
	27.46	412		S1c-3	
30.95	30.40	456	1370	S8-1	
	31.66	475		S8-2	
	30.80	462		S8-3	
31.73	32.93	494	1410	S10 -1	
	30.66	460		S10-2	
	31.60	474		S10-3	
29.86	30.00	450	1350	S4c -1	Group 4
	29.93	449		S4c-2	
	29.66	445		S4c-3	
32.08	31.33	470	1430	S16 -1	
	33.00	495		S16-2	
	31.93	479		S16-3	

Table 7 : Compressive modulus of elasticity of composite mixtures

Average compressive strength (MPa)	Average	Modulus of Elasticity (MPa)	Mix	Group
31.91	9807	9807	S1-2	Group 1
27.11	10297	10297	S2-1	
26.48	3432	3187	S4-1	Group 2
		3677	S4-2	
25.71	2124	1961	S4w-1	
		2451	S4w-2	
		1961	S4w-3	
23.70	3427	3427	S5-3	
28.28	8825	8825	S1c-1	Group 3
30.95	7992.5	7453	S8-1	
		8532	S8-2	
31.73	8090.5	8826	S10 -1	
		7355	S10-2	
29.86	7355	7355	S4c -1	Group 4
32.08	8581	8826	S16 -1	
		8336	S16-2	

Table 8 : Flexural strength results for composite mixtures

Average flexural strength (MPa)	Flexural strength (MPa)	Effective length (L) mm	Flexural load (P) (KN)	Fiber illustration	Mix	Groups
15.34	18.41	450	40.92		S1-1	Group 1
	12.27	450	27.28		S1-2	
11.06	10.8	450	24		s2-1	
	11.32	450	25.17		S2-2	
10.95	10.81	450	24.02		S4-1	Group 2
	11.08	450	24.63		S4-2	
21.99	21.13	450	46.97		S4w -1	
	22.85	450	50.78		S4w -2	
11.88	12.01	450	26.69		S1c-1	Group 3
	11.75	450	26.11		S1c-2	

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






Average flexural strength (MPa)	Flexural strength (MPa)	Effective length (L) mm	Flexural load (P) (KN)	Fiber illustration	Mix	Groups
12.21	12.41	450	27.57		S8-1	Group 4
	12.01	450	26.69		S8-2	
13.81	13.75	450	30.57		S7-1	
	13.86	450	30.82		S7-2	
13.34	13.14	450	29.2		S10 -1	
	13.55	450	30.13		S10-2	
14.15	14.01	450	31.13		S11-1	
	14.30	450	31.78		S11-2	
11.63	11.61	450	25.79		S4c -1	
	11.66	450	25.93		S4c-2	
20.36	20.01	450	44.47		S14 -1	
	20.71	450	46.03		S14-2	
22.07	22.27	400	55.69		S16-1	
	21.87	400	54.69		S16-2	

Table 9 : Flexural modulus of elasticity for composite mix S16

Average	Modulus of elasticity (MPa)	Max load KN	Mix
1519.82	1599.85	55.69	S16-1
	1439.80	54.69	S16-2

Table 10 : Compressive Strength Parallel to Grain Result.

Average	Compressive strength (MPa)	Compressive load (KN)	Mix
36.39	30.31	454.66	S16-1
	33.51	503.43	S16-2
	45.35	680.27	S16-3

Table 11 : Specific gravity and density results.

Density Kg/m ³	Average of specific gravity	Specific gravity	Net weight of sample in water (Kg)	Dry weight of specimen (Kg)	Sample
1314	1.317	1.294	0.017	0.022	S16-1
		1.300	0.030	0.039	S16-2
		1.375	0.016	0.022	S16-3
		1.300	0.010	0.013	S16-4

Table 12 : Coefficient of Linear Thermal Expansion for Mix S16.

Mix S16			
150			Length mm
25			Thickness mm
cold	hot	cold	Temp (c°)
-30	30	-30	
6.52	6.74	6.54	Gage reading mm
contraction		expansion	Δt (C°)
-60		60	
0.22		0.2	ΔL mm
3.6E-03		3.3E-03	ΔL / Δt (mm/C°)
2.4E-05		2.2E-05	Δ (mm/mm.C°)
2.3E-05 mm/mm.C°			CLTE average (Δ)

Table 13 Results of proposed models calculations.

Sleeper	(l) mm	(B) mm	(t) mm	gauge(g)	(L) mm	(qr) KN	Max Pa (Kpa)	Mr (KN.m)	Mc(KN.m)	σ_u Mpa	σ_c Mpa	Tmin (rail seat)	Tmin (center)
T1	2600	230	170	1520	1080	76.51	511.80	10.33	8.42	9.32	7.60	173.03	118.28
T2	2500	230	170	1520	980	76.51	532.27	9.37	10.33	8.46	9.32	160.26	128.48
T3	2400	230	170	1520	880	76.51	554.45	8.42	12.24	7.60	11.05	147.24	137.05
T4	2600	230	160	1520	1080	76.51	511.80	10.33	8.42	10.53	8.58	173.03	118.28
T5	2500	230	160	1520	980	76.51	532.27	9.37	10.33	9.55	10.53	160.26	128.48
T6	2400	230	160	1520	880	76.51	554.45	8.42	12.24	8.58	12.48	147.24	137.05
T7	2600	230	150	1520	1080	76.51	511.80	10.33	8.42	11.98	9.76	173.03	118.28
T8	2500	230	150	1520	980	76.51	532.27	9.37	10.33	10.87	11.98	160.26	128.48
T9	2400	230	150	1520	880	76.51	554.45	8.42	12.24	9.76	14.19	147.24	137.05
T10	2600	220	170	1520	1080	76.51	535.06	10.33	8.42	9.75	7.94	173.03	118.28
T11	2500	220	170	1520	980	76.51	556.46	9.37	10.33	8.85	9.75	160.26	128.48
T12	2400	220	170	1520	880	76.51	579.65	8.42	12.24	7.94	11.55	147.24	137.05
T13	2600	220	160	1520	1080	76.51	535.06	10.33	8.42	11.00	8.97	173.03	118.28
T14	2500	220	160	1520	980	76.51	556.46	9.37	10.33	9.99	11.00	160.26	128.48
T15	2400	220	160	1520	880	76.51	579.65	8.42	12.24	8.97	13.04	147.24	137.05
T16	2600	220	150	1520	1080	76.51	535.06	10.33	8.42	12.52	10.20	173.03	118.28
T17	2500	220	150	1520	980	76.51	556.46	9.37	10.33	11.36	12.52	160.26	128.48
T18	2400	220	150	1520	880	76.51	579.65	8.42	12.24	10.20	14.84	147.24	137.05
T19	2600	210	170	1520	1080	76.51	560.54	10.33	8.42	10.21	8.32	173.03	118.28
T20	2500	210	170	1520	980	76.51	582.96	9.37	10.33	9.27	10.21	160.26	128.48
T21	2400	210	170	1520	880	76.51	607.25	8.42	12.24	8.32	12.10	147.24	137.05

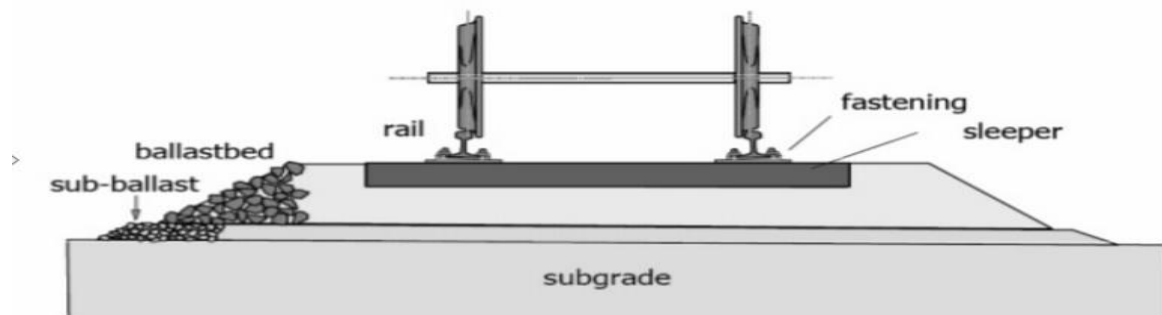


Fig.1. Components of railway track [1].



Fig.2. Types of failure modes, (a) Fungal decay [8], (b) Splitting at ends [7], (c) Spike retention[9]

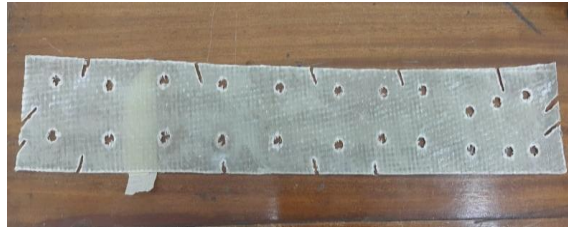


Fig.4. Used laminate in composite mixture



Fig. 5.Cube and Prism Molds [15]



Fig.6. Casting of composite mixture

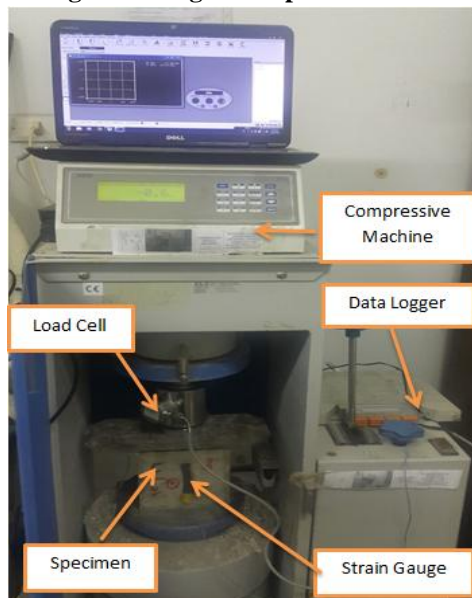


Fig.7. Compressive strength test set up



Fig.8. Actual flexural test

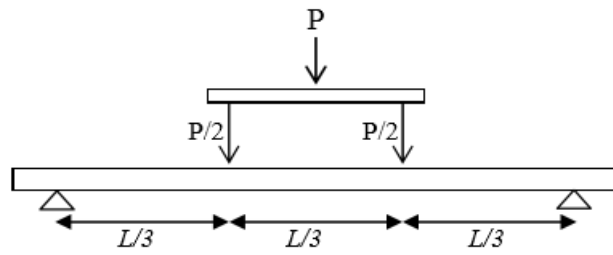


Fig.9. Schematic illustration of flexural test

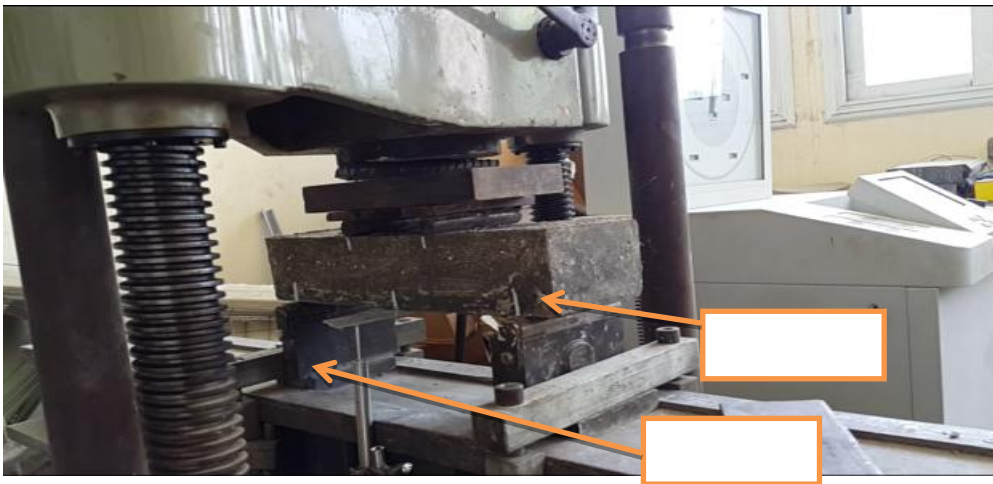


Fig.10. Load -Deflection set up



Fig.11. Specimens of specific gravity test



Fig.12. CLTE test specimen



Fig.13. CLTE set up



Fig.14. Compressive failure mode of specimen S16

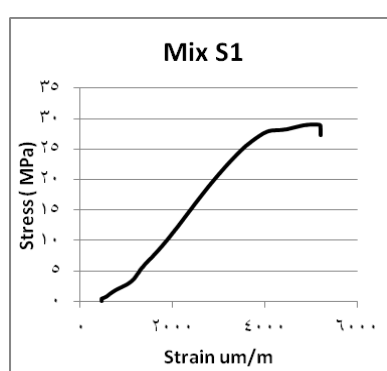


Fig.15. Stress -Strain curve for mix S1

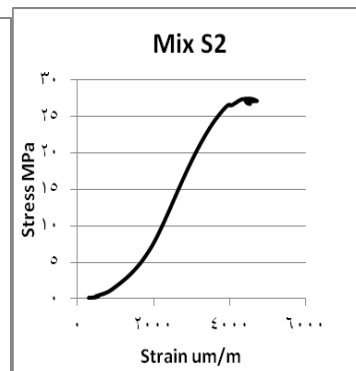


Fig.16. Stress -Strain curve for mix S2

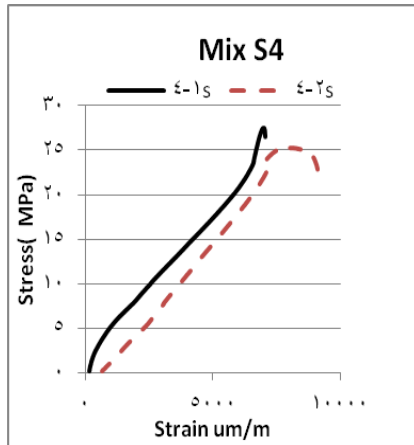


Fig.17. Stress -Strain curve for mix S4

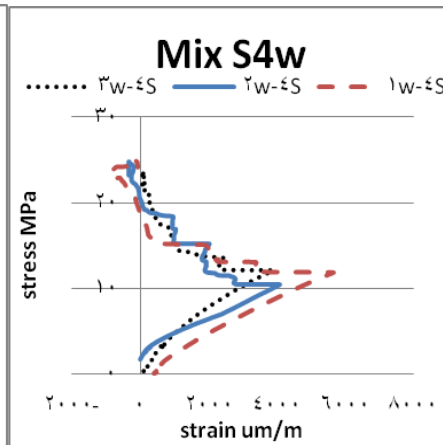


Fig.18. Stress -Strain curve for mix S4w

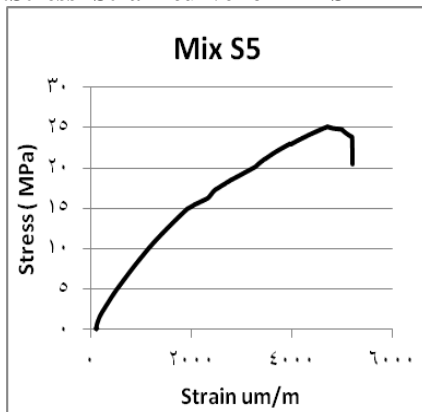


Fig.19. Stress -Strain curve for mix S5

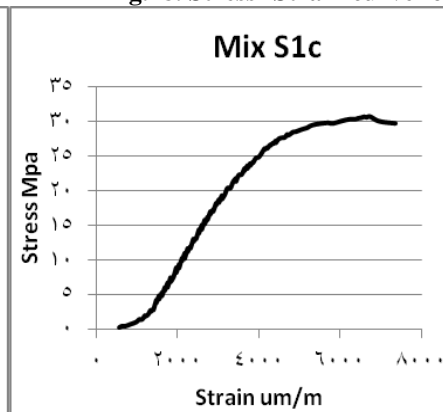


Fig.20. Stress -Strain curve for mix S1C

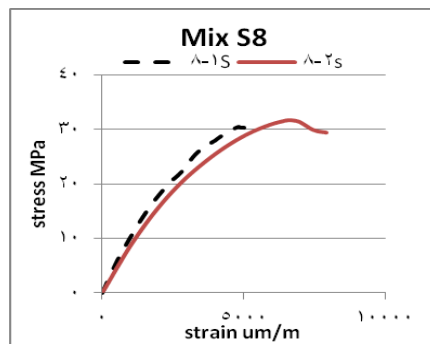


Fig.21. Stress -Strain curve for mix S8

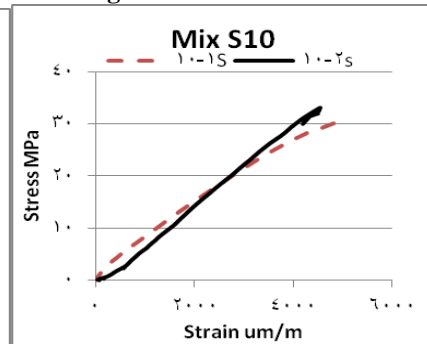


Fig. 22. Stress -Strain curve for mix S10

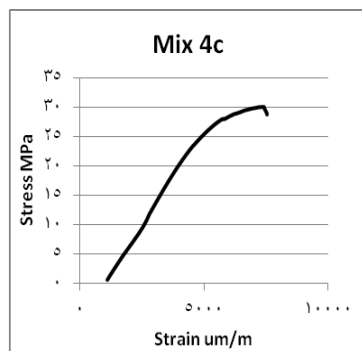


Fig.23. Stress -Strain curve for mix S4

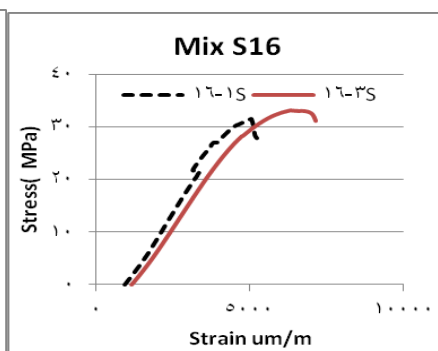


Fig.24. Stress -Strain curve for mix S16

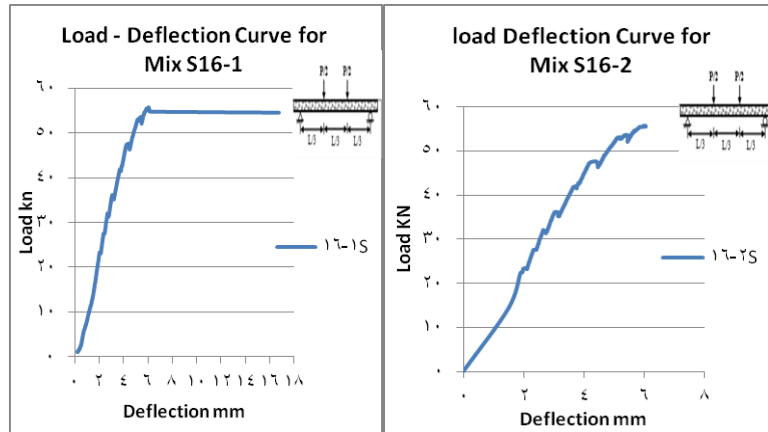


Fig.25. Load- deflection curve for mix S16



Fig.26. Failure mode of specimen S16

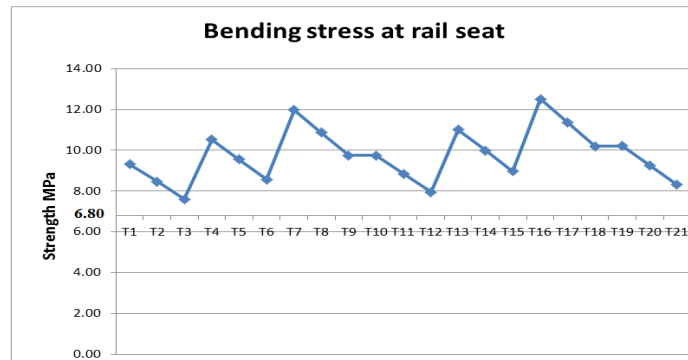


Fig.27 Results of bending stress calculations at rail seat for proposed models

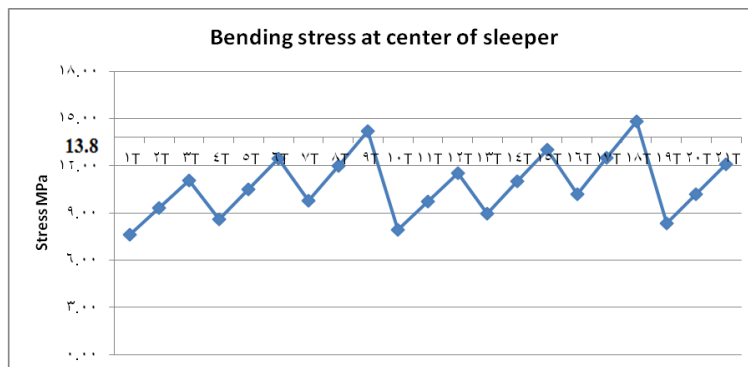


Fig.28 Results of bending stress calculations at the center for proposed model

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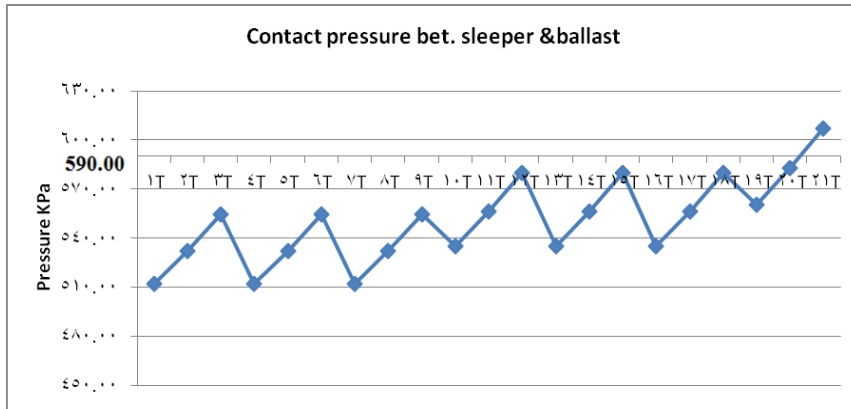


Fig.29 Results of contact pressure calculations between ballast and sleeper for proposed models

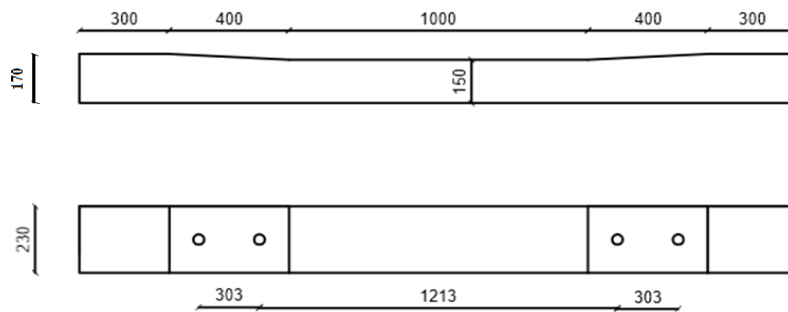


Fig.30 schematic illustration of full scale proposed composite sleeper (all dimensions in mm).



Fig. 31 Actual full scale proposed composite sleeper.