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MATRIX FORMULATION OF CHEBYSHEV SOLUTION TO SHELL PROBLEMS

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ABSTRACT

Any continuous function $f(\xi)$ can be expanded in a Chebyshev series. The n^{th} derivative of the function $f(\xi)$ can be written in matrix form in terms of the expansion coefficients of the function. Also, the product of two functions $f(\xi)$ and $g(\xi)$ can be written in matrix form in terms of the expansion coefficients of the two functions. Therefore, any system of differential equations with variable coefficients can be written as a system of algebraic equations in terms of Chebyshev coefficients of the functions, which can be easily solved. The method is used to solve the problem of isotropic conical shell with different loads and boundary conditions. Results are computed and compared with the exact ones. Comparison proves convergence, accuracy and reliability of the proposed method.

KEYWORDS

Boundary-value problems. Differential equations. Chebyshev series. Shells. Conical shells.

NOMENCLATURE

$$C = \frac{Eh}{1-\nu^2}$$

$$D = \frac{Eh^3}{12(1-\nu^2)}$$

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E	Young's modulus.
M_x, M_θ	longitudinal and tangential bending-stress couples.
N_x, N_θ	longitudinal and tangential normal-stress resultants.
Q_x, Q_θ	longitudinal and tangential shear-stress resultants.
h	shell thickness.
k_x, k_θ	curvature change in the longitudinal and tangential directions
u	displacement of the middle surface in the longitudinal direction.
w	displacement in the direction normal to the middle surface.
α	semi-vertex angle.
ν	Poisson's ratio.
$\varepsilon_x, \varepsilon_\theta$	normal strain in the longitudinal and tangential directions.
$\xi = \frac{x}{L}$	non-dimensional longitudinal coordinate.

INTRODUCTION

Applications of shells or shell-like structures as load-bearing members are common in aerospace, automobile and other industries due to their efficiency as structural components [1]. A considerable amount of research has been done on the development of new shell theories as well as the solution of their equilibrium differential equations. The solution of differential equations has two different approaches: numerical and analytical. Analytical solutions are either exact or approximate.

Exact solutions may be easily obtainable in case of shells with simple geometry and boundary conditions, and uniform thickness and elastic properties. In many circumstances, however, it is not possible to find suitable functions which satisfy both the shell governing differential equations and the geometric and natural boundary conditions [2].

Geckeler [3] suggested an approximate analytical method for the solution of boundary-value problems of thin bending-resistant shells. The Asymptotic Integration method [4] is another approximate analytical method for the solution of shell problems.

Many approximate numerical techniques for the solution of boundary-value problems are available. The finite-difference method [5-7], the finite-element method [8, 9] and numerical-integration methods [10, 11] are examples of these numerical techniques.

A suitable solution function that is able to satisfy many differential equations and boundary conditions is the Chebyshev polynomial. Alwar and Narasimhan [12, 13] used Chebyshev polynomials to solve the problem of spherical shell under axisymmetric and general loads.

The objective of this paper is to reformulate the Chebyshev series technique in matrix form to make it easier, more reliable and less time consuming. Using matrix notation, the function derivatives and function products can be represented in Chebyshev

series in a straightforward and simple manner. This arrangement converts ordinary differential equations with variable coefficients into a simple system of algebraic equations.

Two examples of conical shells with different boundary conditions are worked out in this paper to illustrate the application of the suggested matrix formulation of the Chebyshev technique to shell problems. The first example deals with an isotropic complete cone fixed at its base under uniform internal pressure. The second example deals with an isotropic frustum hinged at its base under concentrated lateral and edge line loads. Results are compared to those obtained by other techniques to demonstrate the accuracy and reliability of the suggested technique.

CHEBYSHEV SERIES REPRESENTATION

Any continuous function $f(\xi)$ in the interval $0 \leq \xi \leq 1$ can be written in Chebyshev series as follows [12, 13]:

$$f(\xi) = \sum_{r=0}^{\infty} a_r T_r(\xi) \tag{1}$$

where:

- + sign means that the 1st term must be halved,
- $a_r \dots$ are constants to be determined so as to obtain the best possible fit.

$$T_r(\xi) = \cos(r t), \quad \cos(t) = 2\xi - 1, \quad 0 \leq \xi \leq 1$$

The shifted Chebyshev polynomials satisfy the recurrence relations:

$$T_{r+1}(\xi) = 2(2\xi - 1)T_r(\xi) - T_{r-1}(\xi), \quad 2 \leq r \leq \infty \tag{2}$$

$$T_0 = 1, \quad T_1 = 2\xi - 1$$

And the orthogonality conditions:

$$\int_0^1 \frac{T_m(\xi)T_n(\xi)}{\sqrt{\xi}\sqrt{1-\xi}} d\xi = \begin{cases} 0 & \text{for } m \neq n \\ \frac{\pi}{2} & \text{for } m = n \neq 0 \\ \pi & \text{for } m = n = 0 \end{cases}$$

For any continuous function $f(\xi)$ the series expansion (1) is fast converging, and a good approximation is obtained by taking a finite number of terms. Therefore, equation (1) is approximated by:

$$f(\xi) = \sum_{r=0}^N a_r T_r(\xi) \tag{3}$$

where, for a known function $f(\xi)$, the coefficients a_r are given by:-

$$a_r = \frac{2}{\pi} \int_0^1 \frac{f(\xi) T_r(\xi)}{\sqrt{\xi} \sqrt{1-\xi}} d\xi \quad 0 \leq r \leq N \quad (4)$$

The first derivative $f'(\xi)$ is expressed in Chebyshev series as [12,13] :

$$f'(\xi) = \sum_{r=0}^{N-1} a_r^{(1)} T_r(\xi) \quad (5)$$

The coefficients $a_r^{(1)}$ satisfy the recursive relation:

$$a_{r-1}^{(1)} - a_{r+1}^{(1)} = 4ra_r \quad , \quad 1 \leq r \leq N \quad (6)$$

Similarly the higher derivatives can be written as:

$$f''(\xi) = \sum_{r=0}^{N-2} a_r^{(2)} T_r(\xi) \quad (7)$$

$$f^{(m)}(\xi) = \sum_{r=0}^{N-m} a_r^{(m)} T_r(\xi)$$

where;

$$a_{r-1}^{(2)} - a_{r+1}^{(2)} = 4ra_r^{(1)} \quad , \quad 1 \leq r \leq N-1$$

$$a_{r-1}^{(m)} - a_{r+1}^{(m)} = 4ra_r^{(m-1)} \quad , \quad 1 \leq r \leq N-(m-1)$$

MATRIX REPRESENTATION OF FUNCTION DERIVATIVES:

The first-order-derivative coefficients $\{a_r^{(1)}\}$ in equation (6) can be written in terms of the original function coefficients $\{a_i\}$ using matrix notation as follows:

$$\{a_r^{(1)}\} = 4 [A] \{a_i\} \quad ; \quad \begin{matrix} r=0, 1, 2, \dots, N-1 \\ i=1, 2, 3, \dots, N \end{matrix} \quad (8)$$

where [A] is an upper triangular matrix of order N x N.

The elements of the matrix a_{ij} are defined as:

$$a_{ij} = \begin{cases} 0 & i > j \quad Or \quad i + j \quad odd \\ j & i \leq j \quad and \quad i + j \quad even \end{cases}$$

The form of [A] for N=5 for example is:

$$A = \begin{bmatrix} 1 & 0 & 3 & 0 & 5 \\ 0 & 2 & 0 & 4 & 0 \\ 0 & 0 & 3 & 0 & 5 \\ 0 & 0 & 0 & 4 & 0 \\ 0 & 0 & 0 & 0 & 5 \end{bmatrix}$$

From equation (8), it is noted that the first-order-derivative coefficients are written in terms of the N coefficients {a_i} (i = 1, 2, ..., N) of the function f(ξ). To represent {a_r⁽¹⁾} in terms of all function coefficients {a_i}, i = 0, 1, 2, ..., N we add a new left column with zero entries in the matrix [A], and the new matrix is termed [A01]. Thus:

$$\{a_r^{(1)}\} = 4 [A01] \{a_i\} \quad ; \quad \begin{matrix} r = 0, 1, 2, \dots, N-1 \\ i = 0, 1, 2, 3, \dots, N \end{matrix} \quad (9)$$

where [A01] is of order N x N+1.

For N=5: [A01] take the form:

$$[A01] = \begin{bmatrix} 0 & 1 & 0 & 3 & 0 & 5 \\ 0 & 0 & 2 & 0 & 4 & 0 \\ 0 & 0 & 0 & 3 & 0 & 5 \\ 0 & 0 & 0 & 0 & 4 & 0 \\ 0 & 0 & 0 & 0 & 0 & 5 \end{bmatrix}$$

The second-order-derivative coefficients {a_r⁽²⁾} in equation (7) can be written in terms of the function coefficients {a_i} using matrix notation as follows:

$$\begin{aligned} \{a_r^{(2)}\} &= 16 [A]_{-1,-1}^{-1} [A01] \{a_i\} \\ \{a_r^{(2)}\} &= 16 [A02] \{a_i\} \quad ; \quad \begin{matrix} r = 0, 1, 2, \dots, N-2 \\ i = 0, 1, 2, \dots, N \end{matrix} \end{aligned}$$

The third-order-derivative coefficients {a_r⁽³⁾} can be written in terms of the function coefficients {a_i} using matrix notation as follows:

$$\begin{aligned} \{a_r^{(3)}\} &= 64 [A]_{-2,-2}^{-1} [A02] \{a_i\} \\ \{a_r^{(3)}\} &= 64 [A03] \{a_i\} \quad ; \quad \begin{matrix} r = 0, 1, 2, \dots, N-3 \\ i = 0, 1, 2, \dots, N \end{matrix} \end{aligned}$$

The general form of the nth-derivative coefficients {a_r⁽ⁿ⁾} can be written in terms of the function coefficients {a_i} using matrix notation as follows:

$$\begin{aligned} \{a_r^{(n)}\} &= (4)^n [A]_{1-n, 1-n}^{-1} [A0(n-1)] \{a_i\} \\ \{a_r^{(n)}\} &= (4)^n [A0n] \{a_i\} \quad ; \quad \begin{matrix} r = 0, 1, 2, \dots, N-n \\ i = 0, 1, 2, \dots, N \end{matrix} \end{aligned}$$

where,

- n ... the order of derivative
- [A]_{1-n, 1-n} ... matrix [A] after deleting the last (n-1) rows and (n-1) columns.
- ⁻¹[] ... matrix [] after deleting the first row.

New Form of Chebyshev Series

Any continuous function f(ξ) in the interval 0 ≤ ξ ≤ 1 and its derivatives can be written in the new matrix form of Chebyshev series as follows:

$$f(\xi) = [T_r] [I] \{a_i\} \quad \begin{matrix} r = 0, 1, 2, \dots, N \\ i = 0, 1, 2, \dots, N \end{matrix} \quad (10)$$

$$f^n(\xi) = (4)^n [T_r] [A0n] \{a_i\} \quad \begin{matrix} r = 0, 1, 2, \dots, N-n \\ i = 0, 1, 2, \dots, N \end{matrix} \quad (11)$$

where; [T_r] is a row matrix whose elements are T_r(ξ). Note that in the new formulation the first term of [T_r] must be halved.

Now consider the general nonhomogenous differential equation of nth order:

$$f^n + f^{n-1} + f^{n-2} + \dots + f' + f = p$$

After expanding each term in Chebyshev series the above differential equation can be written as:

$$\sum_{r=0}^{N-n} a_r^{(n)} T_r(\xi) + \sum_{r=0}^{N-(n-1)} a_r^{(n-1)} T_r(\xi) + \dots + \sum_{r=0}^{N-1} a_r^{(1)} T_r(\xi) + \sum_{r=0}^N a_r T_r(\xi) = \sum_{r=0}^N p_r T_r(\xi) \quad (12)$$

The forcing-function coefficients p_r can be evaluated using equation (4). Equating the coefficients of like Chebyshev polynomial terms on either side, the resulting N+1-n algebraic equations can be written in matrix form using equations (10) and (11) as:

$$[4^{(n)} [A0n] + 4^{(n-1)} [A0(n-1)] + \dots + 4 [A01] + [I]] \{a_i\} = \{p_r\} \quad \begin{matrix} r = 0, 1, 2, \dots, N+1-n \\ i = 0, 1, 2, \dots, N \end{matrix} \quad (13)$$

where all matrices in equation (13) are of the same order (N+1-n x N+1). For all derivatives lower than the highest derivative, the first N+1-n rows are chosen so as to satisfy equation (13). In order to be able to solve equation (13), n additional equations are needed. These additional equations are supplied by the problem boundary conditions.

MATRIX REPRESENTATION OF FUNCTION PRODUCTS

If $f(\xi)$ and $g(\xi)$ are two continuous functions represented by truncated Chebyshev series as:

$$f(\xi) = \sum_{r=0}^N a_r T_r(\xi)$$

$$g(\xi) = \sum_{r=0}^M b_r T_r(\xi)$$

Then the product of these functions can be written in a Chebyshev series as:

$$g(\xi) f(\xi) = \sum_{r=0}^{M+N} c_r T_r(\xi)$$

where

$$c_0 = \sum_{i=0}^{N+M} a_i b_i$$

$$c_r = \frac{1}{2} \sum_{i=0}^{N+M} a_i (b_{i+r} + b_{|i-r|}) \quad ; \quad \begin{matrix} 1 \leq r \leq M+N \\ i+r \leq M \\ |i-r| \leq M \end{matrix} \quad \forall i$$

The $\{c_r\}$ coefficients can be written in terms of the $\{a_i\}$ coefficients only using matrix notation as follows:

$$\{c_r\} = [H] \{a_i\} \tag{14}$$

where

$$r = 0, 1, 2, \dots, N+M$$

$$i = 0, 1, 2, \dots, N$$

$\{c_r\}$... is a column matrix of order $N+M+1 \times 1$

$[H]$... is rectangular matrix of order $N+M+1 \times N+1$

$\{a_i\}$... is a column matrix of order $N+1 \times 1$

The coefficients $\{b_i\}$ are obtained by forcing the function $g(\xi)$ to take on its true values at a number of selected points in the interval $0 \leq \xi \leq 1$. Hence h_{ij} can be written as:

$$h_{ij} = \begin{cases} \frac{1}{4}(b_{|i-j|} + b_{i+j-2}) & \text{for } \begin{matrix} i=1,2,\dots,N+M+1 \\ j=1 \end{matrix} & |i-j| \leq M \\ \frac{1}{2}(b_{|i-j|} + b_{i+j-2}) & \text{for } \begin{matrix} i=1,2,\dots,N+M+1 \\ j=2,3,\dots,N+1 \end{matrix} & i+j-2 \leq M \end{cases} \quad \forall i, j$$

Therefore, equation (14) takes the form:

$$\begin{pmatrix} c_0 \\ c_1 \\ c_2 \\ c_3 \\ \vdots \\ \vdots \\ \vdots \\ c_{N+M} \end{pmatrix} = \frac{1}{2} \begin{bmatrix} b_0 & 2b_1 & 2b_2 & 2b_3 & \dots & \dots \\ b_1 & b_0+b_2 & b_1+b_3 & b_2+b_4 & \dots & \dots \\ b_2 & b_1+b_3 & b_0+b_4 & b_1+b_5 & \dots & \dots \\ b_3 & b_2+b_4 & b_1+b_5 & b_0+b_6 & \dots & \dots \\ \vdots & \vdots & \vdots & \vdots & \dots & \dots \\ \vdots & \vdots & \vdots & \vdots & \dots & \dots \\ \vdots & \vdots & \vdots & \vdots & \dots & \dots \\ \vdots & \vdots & \vdots & \vdots & \dots & \dots \end{bmatrix} \begin{pmatrix} a_0 \\ a_1 \\ a_2 \\ a_3 \\ \vdots \\ \vdots \\ \vdots \\ a_N \end{pmatrix}$$

THE PROBLEM OF AXISYMMETRIC CONICAL SHELL WITH BENDING RESISTANCE:

Equilibrium Equations

The governing equations of a bending-resistant conical shell under normal pressure load q are given by [2]:

$$\begin{aligned} \frac{dN_x}{dx} + \frac{1}{x}(N_x - N_\theta) &= 0 \\ \frac{dQ_x}{dx} + \frac{1}{x}(Q_x - N_\theta \cot \alpha) &= q \\ \frac{dM_x}{dx} + \frac{1}{x}(M_x - M_\theta) - Q_x &= 0 \end{aligned} \tag{15}$$

The shell geometry and stress resultants are shown in Figs. (1) and (2).

Constitutive Relations

The shell constitutive relations between stress resultants and stain and curvature components are given by [2]:

$$\begin{aligned} N_x &= C (\epsilon_x + \nu \epsilon_\theta) & N_\theta &= C (\epsilon_\theta + \nu \epsilon_x) \\ M_x &= D (k_x + \nu k_\theta) & M_\theta &= D (k_\theta + \nu k_x) \end{aligned} \tag{16}$$

Strain-Displacement Relations

The shell strain-displacement relations for small displacements are given by [2]:

$$\begin{aligned} \varepsilon_x &= \frac{du}{dx} & \varepsilon_\theta &= \frac{1}{x}(u + w \cot \alpha) \\ k_x &= -\frac{d^2w}{dx^2} & k_\theta &= -\frac{1}{x} \frac{dw}{dx} \end{aligned} \quad (17)$$

Note that the stress resultants $N_{x\theta}$, $N_{\theta x}$, $M_{x\theta}$, $M_{\theta x}$, the strain components $\varepsilon_{x\theta}$, $k_{x\theta}$ and the displacement component v in the tangential direction are all zero because of the axisymmetric nature of the problem.

Substituting the strain displacement-relations (17) into the stress resultant-strain relations (16), and eliminating Q_x between the last two equations of (15) we end up with the following two equilibrium differential equations:

$$\frac{C}{L^2} \left(\frac{d^2u}{d\xi^2} + \frac{1}{\xi} \frac{du}{d\xi} - \frac{1}{\xi^2} u \right) + \frac{C \cot \alpha}{L^2} \left(\frac{v}{\xi} \frac{dw}{d\xi} - \frac{1}{\xi^2} w \right) = 0 \quad (18)$$

$$\frac{C \cot \alpha}{L^2} \left(\frac{v}{\xi} \frac{du}{d\xi} + \frac{1}{\xi^2} u \right) + \frac{D}{L^4} \left(\frac{d^4w}{d\xi^4} + \frac{2}{\xi} \frac{d^3w}{d\xi^3} - \frac{1}{\xi^2} \frac{d^2w}{d\xi^2} + \frac{1}{\xi^3} \frac{dw}{d\xi} + \frac{C \cot^2 \alpha}{D} \frac{1}{\xi^2} w \right) = p$$

Expanding $u(\xi)$ and $w(\xi)$ in $(N+1)$ -term Chebyshev series we have a total of $2N+2$ unknown coefficients. The functions $\frac{1}{\xi}$, $\frac{1}{\xi^2}$ and $\frac{1}{\xi^3}$ appearing in the equilibrium equations are also expanded in Chebyshev series having $M+1$ terms. The $M+1$ expansion coefficients can be computed easily by forcing the functions $\frac{1}{\xi}$, $\frac{1}{\xi^2}$ and $\frac{1}{\xi^3}$

to take on their actual values at a number of chosen points in the interval $0 \leq \xi \leq 1$. Using matrix notation for the functions and function derivatives, and applying the rule of matrix multiplications, the equilibrium equations can be written as a system of algebraic equations in the following matrix form:

$$\begin{aligned} \frac{C}{L^2} [16[A02] + 4[H101] - [H2]] \{u_i\} + \frac{C \cot \alpha}{L^2} [4v[H101] - [H2]] \{w_i\} &= \{0\} \\ \frac{C \cot \alpha}{L^2} [4v[H101] + [H2]] \{u_i\} + \frac{D}{L^4} [256[A04] + 128[H103] - 16[H202] + 4[H301] + & \\ + \frac{CL^2 \cot^2 \alpha}{D} [H2]] \{w_i\} &= \{p_i\} \end{aligned} \quad (19)$$

where:

[H1] ... is the matrix of coefficients of the function $\frac{1}{\xi}$

[H2] ... is the matrix of coefficients of the function $\frac{1}{\xi^2}$

[H3] ... is the matrix of coefficients of the function $\frac{1}{\xi^3}$

$$\begin{aligned} [H101] &= [H1] [A01] \\ [H103] &= [H1] [A03] \\ [H202] &= [H2] [A02] \\ [H301] &= [H3] [A01] \end{aligned}$$

The highest derivative in the first of equations (19) is of order 2, so the number of algebraic equations is N-1 equations. The highest derivative in the second of equations (19) is of order 4, so the number of algebraic equations is N-3 equations. The total number of algebraic equations is 2N-4 along with 6 boundary conditions at $\xi=0$ and $\xi=1$, leading to 2N+2 equations in 2N+2 unknowns, which can be easily solved.

It is finally important to note that all matrices in the first of equations (19) are of order (N-1 x N+1), while all matrices in the second of equations (19) are of order (N-3 x N+1).

Boundary Conditions

Fortunately, it is relatively easy to represent any shell boundary conditions for the functions expanded in Chebyshev series.

Top-vertex conditions

At the vertex $\xi=0$:

1. The displacement perpendicular to the axis of the shell is zero because of axisymmetry, which leads to:

$$u \sin \alpha + w \cos \alpha = 0$$
2. $\frac{dw}{d\xi} = 0$ for finite M_x and M_θ
3. The vertical displacement ($-u \cos \alpha + w \sin \alpha$) at the vertex is a maximum, which leads to: $\frac{du}{d\xi} = 0$

Clamped-edge conditions

1. $u = 0$
2. $w = 0$
3. $\frac{dw}{d\xi} = 0$

Hinged-edge conditions:

1. $u = 0$
2. $w = 0$
3. $M_x = 0$ i.e. $\frac{d^2w}{d\xi^2} + \frac{\nu}{\xi} \frac{dw}{d\xi} = 0$

The previous sets of boundary conditions can be written in matrix form by calculating the function or its derivatives at $\xi=0$ or $\xi=1$ as follow:

Vertex boundary conditions:

At $\xi = 0$

1. $\sin \alpha [TR0] \{u_i\} + \cos \alpha [TR0] \{w_i\} = 0$
2. $4 [TR01] [A01] \{w_i\} = 4 [T0A01] \{w_i\} = 0$
3. $4 [TR01] [A01] \{u_i\} = 4 [T0A01] \{u_i\} = 0$

where:

[TR0] is a row matrix of N+1 Chebeychev terms at $\xi = 0$

[TR01] is a row matrix of N Chebeychev terms at $\xi = 0$

Note: the first term of Chebeyshev terms must be halved.

Clamped-base boundary conditions:

At $\xi = 1$

1. $[TR1] \{u_i\} = 0$
2. $[TR1] \{w_i\} = 0$
3. $4 [TR11] [A01] \{w_i\} = 4 [T1A01] \{w_i\} = 0$

where:

[TR1] is a row matrix of N+1 Chebeychev terms at $\xi = 1$

[TR11] is a row matrix of N Chebeychev terms at $\xi = 1$

Note: the first term of Chebeyshev terms must be halved.

Hinged-edge conditions:

At $\xi = 1$

1. $[TR1] \{u_i\}$
2. $[TR1] \{w_i\} = 0$
3. $(16 [TR12] [A02] + 4 [TR11] [A01]) \{w_i\} = (16 [T1A02] + 4 [T1A01]) \{w_i\} = 0$

where:

[TR11] is a row matrix of N Chebeychev terms at $\xi = 1$

[TR12] is a row matrix of N-1 Chebeychev terms at $\xi = 1$

Note: the first term of Chebeyshev terms must be halved.

RESULTS AND DISCUSSIONS

Problem 1:

A complete cone with clamped base under uniform internal pressure p, Fig.3.

Material:

$E = 30 \cdot 10^6$ psi (206.85 GPa);

$\nu = .3$

Geometry:

$L/h = 50$;

$\alpha = 45$

The functions $\frac{1}{\xi}$, $\frac{1}{\xi^2}$ and $\frac{1}{\xi^3}$ appearing in equation (18) are forced to take on their actual values at 10 points other than $\xi=0$. The system of equations(19), (20) and (21) are solved and the coefficients $\{u_i\}$ and $\{w_i\}$ are obtained. The strain and curvature components ϵ_0 and k_0 are computed at $\xi=0.001$ to avoid the singularity at $\xi=0$.

Table 1 shows the results of a convergence study with regard to $\overline{N_x}$ ($2N_x/p \tan \alpha$) and $\overline{N\theta}$ ($2 N\theta/p \tan \alpha$) along the generator. It can be seen that the solution convergence is good, and that fairly accurate results can be obtained with 20 terms in the Chebyshev series. Table 2 and Figs. 5-8 compare the computed results $\overline{N_x}$, $\overline{N\theta}$, $\overline{M_x}$ ($200 M_x / p L^2 \tan (\alpha)$) and $\overline{M\theta}$ ($200 M_\theta / p L^2 \tan (\alpha)$) using 18 terms with the exact results obtained in reference [2] using Kelvin functions. It is seen that the computed results are very close to the exact ones with an average error less than 1.85 %

Problem 2:

A truncated frustum under a line load normal to the surface, and a horizontal line load at the base, Fig. 4. The upper edge is free, while the lower base is supported on rollers.

Material:

E=200 Gpa v=.3

Geometry:

L=1 m L1 = 0.4226497 m Lp=0.5381198 m $\alpha=30^\circ$

Applied load:

P = 1000 N/m H_t = -100 N/m M_x = 10 Nm/m

A new non-dimensional parameter $\xi = \frac{x-L_1}{L}$ is used to make the upper edge lie at $\xi=0$.

The system of equations (18) is applied except that the functions $\frac{1}{\xi}$, $\frac{1}{\xi^2}$ and $\frac{1}{\xi^3}$ become $\frac{1}{(\xi + \frac{L_1}{L})}$, $\frac{1}{(\xi + \frac{L_1}{L})^2}$ and $\frac{1}{(\xi + \frac{L_1}{L})^3}$ respectively. Matrices [H1], [H2] and [H3] appearing in these equations are calculated for the new functions. The line load P is

distributed over $\Delta\xi = \xi_2 - \xi_1$ to convert it into uniform pressure. Equation (4) is used to obtain the loading coefficients $p_r = \frac{2}{\pi} \int_{\xi_1}^{\xi_2} \frac{f(\xi)T_r(\xi)}{\sqrt{\xi}\sqrt{1-\xi}} d\xi$; $0 \leq r \leq N$.

Boundary conditions:

Free-edge conditions:

Roller-supported-edge conditions:

At $x = L_1$ i.e. $\xi = 0$

At $x = L_2$ i.e. $\xi = 1$

- 1- $N_x = 0$
- 2- $Q_x = 0$
- 3- $M_x = 10$

- 1- Vertical displacement ($-u \cos \alpha + w \sin \alpha$) = 0
- 2- $M_x = 0$
- 3- Horizontal force ($N_x \sin \alpha + Q_x \cos \alpha$) = -100

The results are computed and compared with the exact solution obtained in reference [14] using Green's function. Table 3 and Figs. (9-12) compare the distributions of the normal displacement w and the stress resultants N_x , M_x , and Q_x . The normal displacement w and the axial force N_x are calculated using 20 Chebebyshev terms, while the transverse shear Q_x and Longitudinal moment M_x are calculated using 70 terms because of the discontinuity at the point of load application. The table and figures show good agreement between the results with small discrepancy around the singularity.

CONCLUSION

A technique is presented for the solution of boundary-value problems described by a system of simultaneous differential equations with variable coefficients by expressing the unknown functions in terms of Chebebyshev series. A new matrix formulation is presented for the technique, which systematically transforms the problem into a system of algebraic equations, which can be readily solved in a short time on the computer.

The suggested technique is applied to two problems of thin, bending-resistant conical shells with different boundary conditions and external loads. Results are compared with the exact ones, and good agreement is found between the results. The applications prove that the suggested technique is accurate and easily applicable to difficult boundary-value problems.

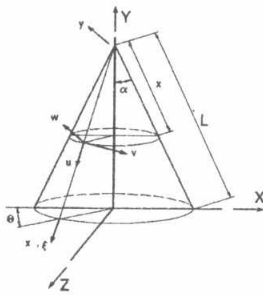


Fig. 1 Conical Shell Geometry

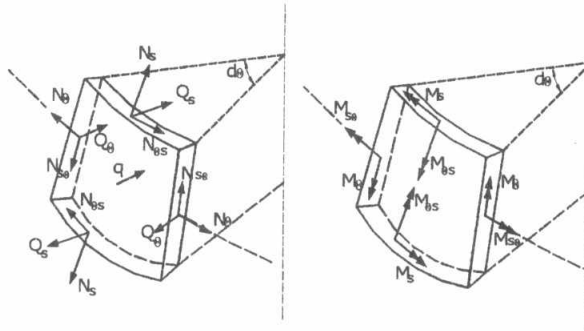


Fig. 2 Shell Stress Resultants

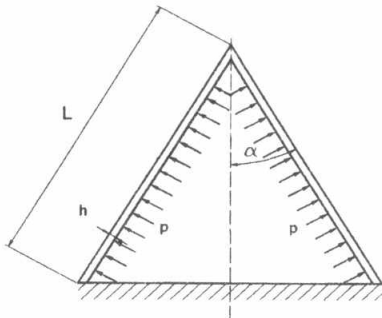


Fig. 3 Pressurized conical shell with clamped edge

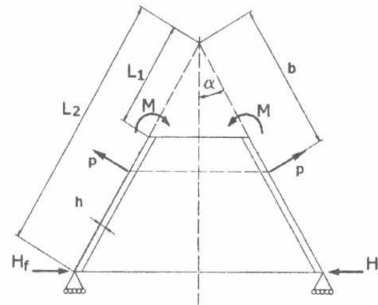


Fig. 4 Truncated cone under line loads

Table 1 Convergence Study Of Conical Shell Clamped At Its Base.

\overline{Nx}									
ξ	0.2	0.3	0.4	0.5	0.6	0.7	0.8	0.9	1
16	0.1988	0.3034	0.399	0.5	0.602	0.715	0.815	0.857	0.823
18	0.2037	0.2996	0.401	0.498	0.604	0.713	0.816	0.857	0.823
20	0.2030	0.3014	0.399	0.499	0.603	0.714	0.817	0.857	0.823

$\overline{N\theta}$									
16	0.3850	0.617	0.777	1.012	1.252	1.491	1.471	0.857	0.246
18	0.4025	0.5957	0.798	0.999	1.254	1.495	1.467	0.858	0.246
20	0.3994	0.6004	0.794	1.002	1.252	1.496	1.467	0.858	0.246

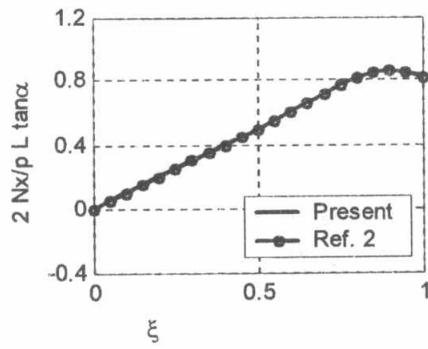


Fig. 5 Axial-force distribution

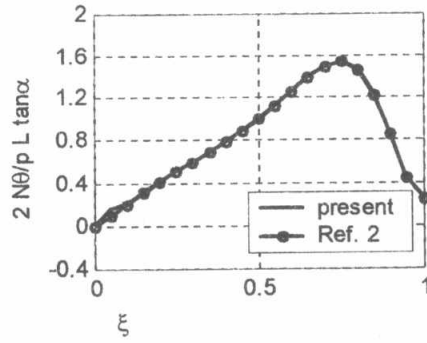


Fig. 6 Circumferential-force distribution

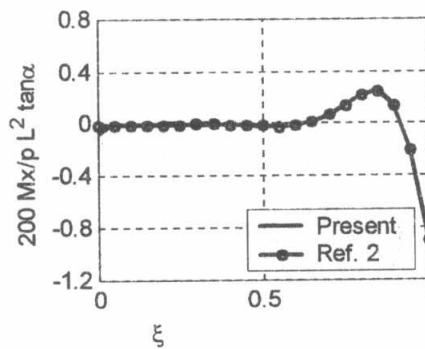


Fig. 7 Axial-moment distribution

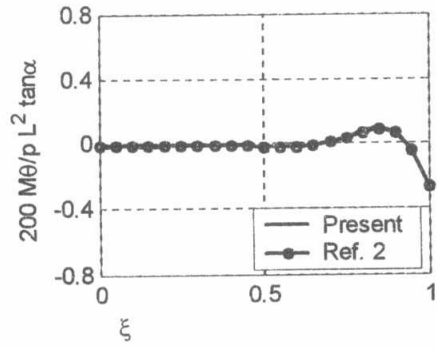


Fig. 8 Circumferential-moment distribution

Table 2 Comparison of computed results (N=20) of complete cone with the exact ones.

ξ	\bar{N}_x		\bar{N}_θ		\bar{M}_x		\bar{M}_θ	
	Present	Exact	Present	Exact	Present	Exact	Present	Exact
0.2	0.2031	0.2000	0.3995	0.4003	-0.0134	-0.0143	-0.0143	-0.0143
0.3	0.3014	0.3001	0.6004	0.5995	-0.0132	-0.0134	-0.0138	-0.0139
0.4	0.3999	0.3994	0.7941	0.7948	-0.0161	-0.0153	-0.0146	-0.0144
0.5	0.4997	0.4987	1.0024	1.0020	-0.0246	-0.0255	-0.0183	-0.0185
0.6	0.6034	0.6028	1.2525	1.2530	-0.0241	-0.0231	-0.0199	-0.0196
0.7	0.7141	0.7143	1.4966	1.4964	0.0562	0.0565	0.0075	0.0076
0.8	0.817	0.8149	1.4677	1.4635	0.2154	0.2146	0.0717	0.0717
0.9	0.8573	0.8585	0.8589	0.8573	0.1341	0.1343	0.0685	0.0684
1.0	0.8233	0.8207	0.2470	0.2462	-0.8936	-0.8968	-0.2681	-0.2690

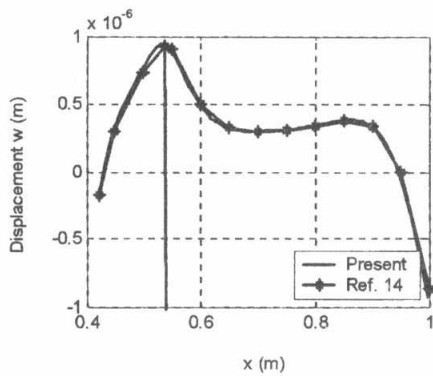


Fig. 9 Normal displacement w (frustum)

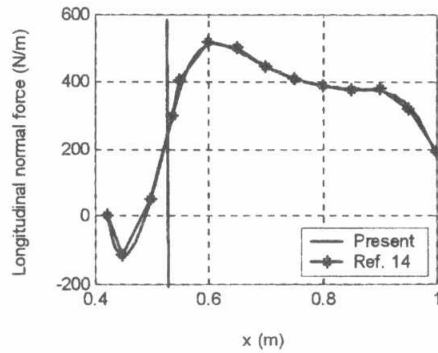


Fig. 10 Axial force distribution (frustum)

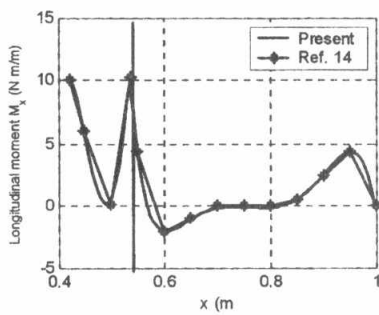


Fig. 11 Bending-moment distribution (frustum)

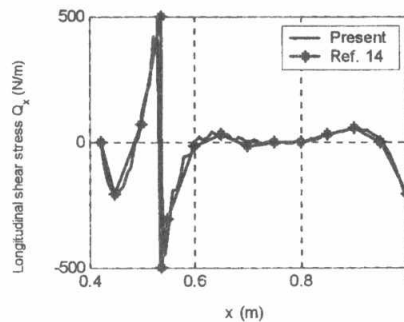


Fig. 12 Shear-force distribution (frustum)

Table 3 Comparison of computed results ($N=70$) of a frustum with the exact ones.

ξ	$W \times 10^{-6}$		Nx		Mx		Qx	
	Present	Exact	Present	Exact	Present	Exact	Present	Exact
0.6	0.4921	0.3427	509.94	517.89	-2.1947	-2.096	-13.48	-13.92
0.7	0.3017	0.3258	446.58	442.72	-0.1805	-0.149	1.34	1.41
0.8	0.3462	0.3469	388.72	390.71	-0.1202	-0.181	2.91	2.764
0.9	0.3384	0.3386	373.15	374.56	2.4440	2.453	49.07	50.84
1.0	-0.9361	-0.937	182.149	182.99	-0.0000	0.001	-220.63	-221.15

REFERENCES

1. Timoshenko, S. P. and Woinowsky-Krieger, S. Theory of plates and shells, 2nd ed., Mc-Graw Hill Book Co., New York, 1959
2. Harry Kraus, Thin Elastic Shells, John Wiley & sons, Inc., New York. London. Sydney, 1967.
3. Geckeler, J. W., "Über die Festigkeit achensymmetrischer Schalen," Forschungsarb. Ingwes., Berlin, 276, 1-52 (1926).
4. Langer, R. E., "On the Asymptotic Solution of Ordinary Differential Equations." Trans. Am. Math. Soc., 37, 397-416 (1935).
5. Forsythe, G. E., and Wasow, W. R., Finite Difference Methods for Partial Differential Equations, New York: John Wiley (1960).
6. Penny, R. K., "Symmetric Bending of the General Shell of Revolution by Finite Difference Methods, J. Mech. Eng. Sci., 3, 369-377 (1961)
7. Hubka, R. E., "A Generalized Finite Difference Solution of Axisymmetric Stress States in Thin Shells of Revolution," Report EM-11-19, Space Technology Laboratories Los Angeles, Calif. (1961).
8. Meyer, R. R., and Harmon, M. B., "Conical Segment Method for Analyzing Open Crown Shells of Revolution for Edge Loading," AIAA J., 1, 886-891 (1963)
9. T. Kant and M. P. Menon, Estimation of interlaminar stresses in fiber reinforced composite cylindrical shells, Computers & Structures. Vol. 38, No. 2, pp. 131-147, 1991.
10. Galletly, G. D., " Influence Coefficients for Open Crown Hemispheres," Trans. A. S. M. E., 82A, 73-81 (1960).
11. Kalnins, A., " Analysis of Shells of Revolution Subjected to Symmetrical and Nonsymmetrical Loads," J. Appl. Mech., 31, 467-476 (1964)
12. R. S. Alwar and M. C. Narasimhan, Application of Chebyshev polynomials to the analysis of laminated axisymmetric spherical shell. Comps. Struct. 15, 215-237, 1990.
13. R. S. Alwar and M. C. Narasimhan, Analysis of laminated orthotropic spherical shells subjected to asymmetric loads. Computer & Structures vol. 41, No. 4, pp. 611-620, 1991.
14. S. A. Tavares, Thin conical shells with constant thickness and under axisymmetric load. Computer & Structures vol. 60, No. 6, pp. 895-921, 1996.