

MILITARY TECHNICAL COLLEGE CAIRO - EGYPT

AXISYMMETRIC VIBRATION OF STEPPED

ANNULAR PLATES

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ABSTRACT

Free axisymmetric vibrations of annular stepped plates have been studied on the basis of the classical theory of plates. Applying the uniform plate solution to each zone of different thickness; considering the continuity and boundary conditions, yields a system of eight equations. Frequency determinants of such plates, with different step radii, are derived for various combinations of boundary conditions. Consequently the values of the resonant frequencies of the first two symmetric modes are calculated. Different graphs are included to facilitate the use of this material for design purpose.

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KEYWORDS

Transvers vibration, annular plates, stapped circular plates, boundary conditions, natural frequency, symmetric modes.

INTRODUCTION

The vibration analysis of circular plates with uniform thickness is a classical problem which can be found in literature. In a recent papers, authors like Kutter [1], Kirkhope [2] and others have studied the vibration of circular plates with variable thickness. Some of these plates which are of interest to engineers are the annular plates. The vibration of these plates with uniform thickness has been treated by Vogel [3] and Raju [4], while the vibration of annular plates with variable thickness was studied by authors such as Soni [5] and Irie [6].

The previous two papers [7] and [8] were performed to obtain the modal analysis of stepped circular plates with different boundary conditions and stepped plates with elastically built-in boundary condition respectively.

In the present work, the modal analysis of symmetric vibration of stemped annular plates is developed to describe the effect of the step radius on the natural frequencies of these plates. The results are obtained for plates with different combinations of boundary conditions. Fortunantely the solution procedure gave good results.

PROBLEM FORMULATION

In this work, the vibration is assumed to be governed by small deflection, thin-plate theory, in which the effect of rotary inertia and shear deformation are ignored. The differential equation govering the transverse vibration of the circular plates in polar coordinate 181 is

$$\nabla^4 w(r) - \beta^4 w(r) = 0 \tag{1}$$

where

$$\beta' = 12(1 - \gamma^2) \rho \omega^2 / Eh^2$$
 (2)

The solution of eqn.(1) is

$$W(r) = A J (\beta r) + B Y (\beta r) + C I (\beta r) + F K (\beta r)$$
 (3)

The values of the constants A,B,C,F and β are determined from the boundary conditions of the plates.

Figure 1 shows the cross section of two particular forms of annular plates with stepped thickness. The plate may be described by two zones I and II in which each one represents annular plate with uniform thickness. The common radius r_{\uparrow} is called the step radius.

For the two annular plates I and II, the solutions are given by eqn. (3) respectively

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$$W_{1}(r) = A_{1} J_{0}(\beta_{1}r) + B_{1} Y_{0}(\beta_{1}r) + C_{1} I_{0}(\beta_{1}r) + F_{1} K_{0}(\beta_{1}r)$$
(4)

$$W_{2}(r) = A_{2}J_{0}(\beta_{2}r) + B_{2}Y_{0}(\beta_{2}r) + C_{2}I_{0}(\beta_{2}r) + F_{2}K_{0}(\beta_{2}r)$$
(5)

$$\beta_1^4 = 12(1 - v^2) \rho \omega^2 / Eh_1^2$$
 (6)

$$\beta_2^4 = 12(1 - v^2) \rho \omega^2 / Eh_2^2$$
 (7)

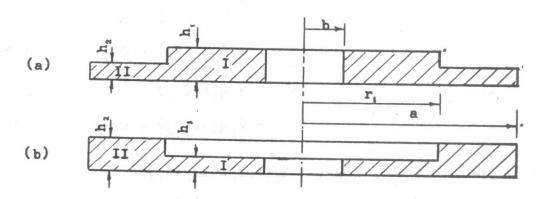


Fig.1 Cross section of two types of stepped annular plates.

a- plate with central region raised.b- plate with central region unraised.

The different coefficients of eqns. (4) and (5) must be chosen to satisfy the continuity conditions at the step radius r_i and the boundary conditions at r=a.

CONTINUITY CONDITIONS

The following are the four continuity conditions for each of deflection, slope, shear force and bending moment respectively

$$W_1(\mathbf{r}_i) = W_2(\mathbf{r}_i) \tag{8}$$

$$dW_{1}(r_{1})/dr = dW_{2}(r_{1})/dr$$
 (9)

$$Q_{1}(\mathbf{r}_{1}) = Q_{2}(\mathbf{r}_{1}) \tag{10}$$

$$H_{r_1}(r_1) = H_{r_2}(r_1)$$
 (11)

Where Q and M are the expressions for shear force and bending moment respectively, [9]

The expressions of W_1 and W_2 will have the following forms at $r_{\rm E}r_1$.

$$W_{1}(\mathbf{r}_{1}) = A_{1}J_{0}(\lambda_{1}) + B_{1}Y_{0}(\lambda_{1}) + C_{1}I_{0}(\lambda_{1}) + F_{1}K_{0}(\lambda_{1})$$
(12)



Consequently the condition (8) can be expressed as follows:

$$A_{1} J_{0}(\lambda_{1}) + B_{1} Y_{0}(\lambda_{1}) + C_{1} I_{0}(\lambda_{1}) + F_{1} K_{0}(\lambda_{1})$$

$$-A_{2} J_{0}(\lambda_{2}) + B_{2} Y_{0}(\lambda_{2}) + C_{2} I_{0}(\lambda_{2}) - F_{2} K_{0}(\lambda_{2}) = 0$$
(14)

Referring to the differentiation of the Bessel's functions [10], the conditions (9), (10) and (11) are given respectively as:

$$A_{1} \beta_{1} J_{1} (\lambda_{1}) + B_{1} \beta_{1} Y_{1} (\lambda_{1}) - C_{1} \beta_{1} I_{1} (\lambda_{1}) + F_{1} \beta_{1} K_{1} (\lambda_{1})$$

$$-A_{2} \beta_{2} J_{1} (\lambda_{2}) - B_{2} \beta_{2} Y_{1} (\lambda_{2}) + C_{2} \beta_{2} I_{1} (\lambda_{2}) - F_{2} \beta_{2} K_{1} (\lambda_{2}) = 0,$$
(15)

$$A_{1} q_{1} J_{1} (\lambda_{1}) + B_{1} q_{1} Y_{1} (\lambda_{1}) + C_{1} q_{1} I_{1} (\lambda_{1}) - F_{1} q_{1} K_{1} (\lambda_{1})$$

$$-A_{2} q_{2} J_{1} (\lambda_{2}) - B_{2} q_{2} Y_{1} (\lambda_{2}) - C_{2} q_{2} I_{1} (\lambda_{2}) + F_{2} q_{2} K_{1} (\lambda_{2}) = 0,$$
(16)

$$A_{1} m_{1} [-\beta_{1} J_{0}(\lambda_{1}) + LJ_{1}(\lambda_{1})] - B_{1} m_{1} [-\beta_{1} Y_{0}(\lambda_{1}) - LY_{1}(\lambda_{1})]$$

$$+C_{1} m_{1} [-\beta_{1} I_{0}(\lambda_{1}) - LI_{1}(\lambda_{1})] + F_{1} m_{1} [-\beta_{1} K_{0}(\lambda_{1}) + LK_{1}(\lambda_{1})]$$

$$-A_{2} m_{2} [-\beta_{2} J_{0}(\lambda_{2}) + LJ_{1}(\lambda_{2})] - B_{2} m_{2} [-\beta_{2} Y_{0}(\lambda_{2}) + LY_{1}(\lambda_{2})]$$

$$-C_{2} m_{2} [-\beta_{2} I_{0}(\lambda_{2}) - LI_{1}(\lambda_{2})] - F_{2} m_{2} [-\beta_{2} K_{0}(\lambda_{2}) + LK_{1}(\lambda_{2})] = 0$$
(17)

where

$$\lambda_{1} = \beta_{1} r_{1}, \qquad \lambda_{2} = \beta_{2} r_{1}
q_{1} = \beta_{1}^{3} h_{1}^{3}, \qquad q_{2} = \beta_{2}^{3} h_{2}^{3}
m_{1} = \beta_{1} h_{1}^{3}, \qquad m_{2} = \beta_{2} h_{2}^{3}$$

$$L = (1 - v)/r.$$
(18)

BOUNDARY CONDITIONS

The boundaries of the annular plate either at r=a or r=b may be clamped, simply supported or free. By using the expressions of the deflection described in eqns.(12) and (13) and their derivatives we can obtain the following forms of the boundary conditions for each case.

Clamped Supported Conditions

$$a-W=0$$
 or

$$A_{i} J_{0}(\lambda_{i}) + B_{i} Y_{0}(\lambda_{i}) + C_{i} I_{0}(\lambda_{i}) + F_{i} K_{0}(\lambda_{i}) = 0$$
 (19)

b-dW/dr=0, or

$$-A_{i}J_{i}(\lambda_{i}) - B_{i}Y_{i}(\lambda_{i}) + C_{i}I_{i}(\lambda_{i}) - F_{i}K_{i}(\lambda_{i}) = 0$$
 (20)

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Simply Supported Conditions

a-W=0, see eqn.(19)

b-M=0 or

$$A_{i} \begin{bmatrix} -\beta_{i} J_{0}(\lambda_{i}) + PJ_{i}(\lambda_{i}) \end{bmatrix} + B_{i} \begin{bmatrix} -\beta_{i} Y_{0}(\lambda_{i}) + PY_{i}(\lambda_{i}) \end{bmatrix}$$

$$+C_{i} \begin{bmatrix} \beta_{i} I_{0}(\lambda_{i}) - PI_{i}(\lambda_{i}) \end{bmatrix} + F_{i} \begin{bmatrix} \beta_{i} K_{0}(\lambda_{i}) + PK_{i}(\lambda_{i}) \end{bmatrix} = 0$$
(21)

Free Supported Conditions

a = Q = 0 or

$$A_{i}J_{i}(\lambda_{i}) + B_{i}Y_{i}(\lambda_{i}) + C_{i}I_{i}(\lambda_{i}) = F_{i}K_{i}(\lambda_{i}) = 0$$
 (22)

b = M = 0, see eqn.(21)

where

$$\lambda_{i} = \beta_{i} r,$$

$$i = \begin{cases} 1 & \text{for the inner boundary} \\ 2 & \text{for the outer boundary}, \end{cases}$$

$$r = \begin{cases} b & \text{for } i = 1 \\ a & \text{for } i = 2, \end{cases}$$

$$P = (1 - v)/a$$
(23)

Hence, it is clear that for an annular plate, there are four continuity conditions and four boundary conditions. This system of equations can be arranged in a matrix form in order to obtain the frequency determinant as follows:

$$[Y] [C] = [0]$$
 (24)

where

[Y] is 8x8 square matrix

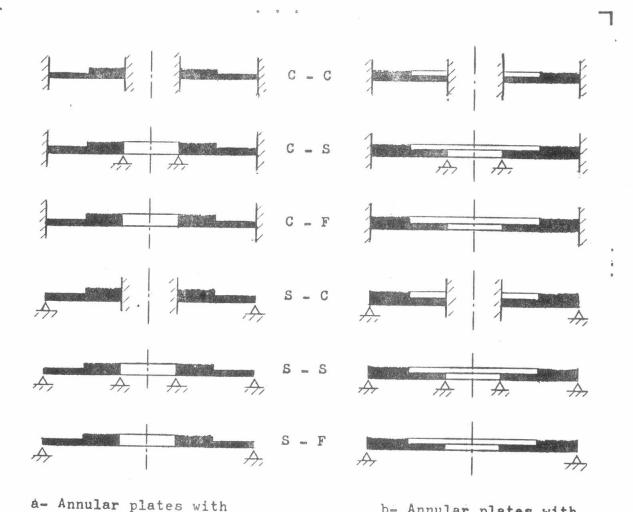
[C] is the column matrix containing the coefficients described before in eqns. (4) and (5).

The natural frequencies of the symmetric modes of such plates can be obtained by equating the determinant of the matrix [Y] by zero.

COMPUTED RESULTS AND DISCUSSIONS

The numerical computation for the natural frequencies are obtained for annular plates with different combinations of the boundary conditions. Figure 2 shows such plates in which the inner radius b=45 mm and the outer radius a=150 mm. In the case of plates with central region raised, the thicknesses h, and h₂ are 1.0 and 0.75 mm respectively, while in the case of plates with central region unraised, h, and h₂ are 0.75 and 1.0 mm respectively.





central region raised.

b- Annular plates with central region unraised.

Fig.2 Stepped annular plates with different boundary conditions.

These models are made of Pirespex with Young's modulus of 262 x10 N.m., Poisson's ratio v = 0.38 and a density of 1.237 x10 Kg.m..

The results obtained by one of the computer methods are expressed in dimensionless form, therefore there are eight step radius ratios r/a vary from 0.3 to 1.0 which are corresponding to the step radii $r_1 = 45$, 60,75,90,105,120,135 and 150 mm. Also the natural frequencies, f are expressed in dimensionless form f/f, where f represents the natural frequency of uniform annular plate with the smallest thickness 0.75 mm either in the case of central region raised or unraised.

Figure 3 shows the results of natural frequencies of C-F, C-S and C-C stepped annular plates, while Fig. 4 shows the results for S-F, S-S and S-C stepped annular plates. The two figures show the effect of the variation of step radius on the natural frequencies of the different plates at the first symmetric mode OO. Similarly, Figs. 5 and 6 show the same relationships at the second symmetric mode O1.

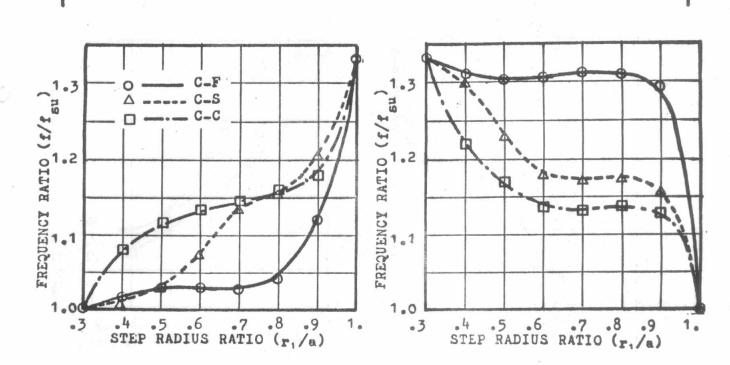
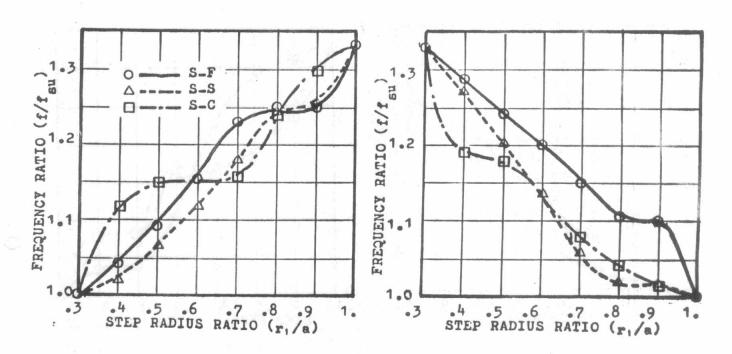


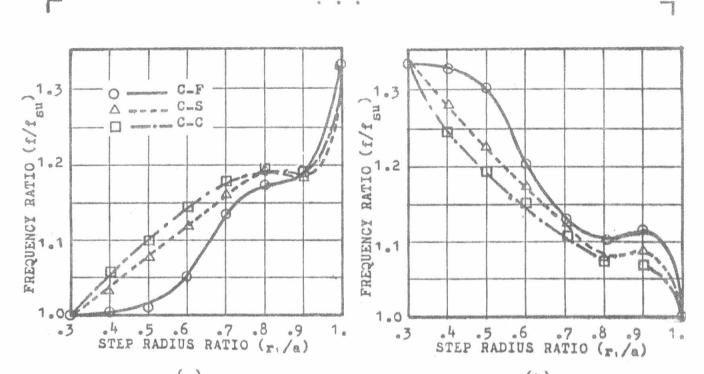
Fig.3 Natural frequencies of C-F, C-S and C-C stepped annular plates at the first symmetric mode 00.



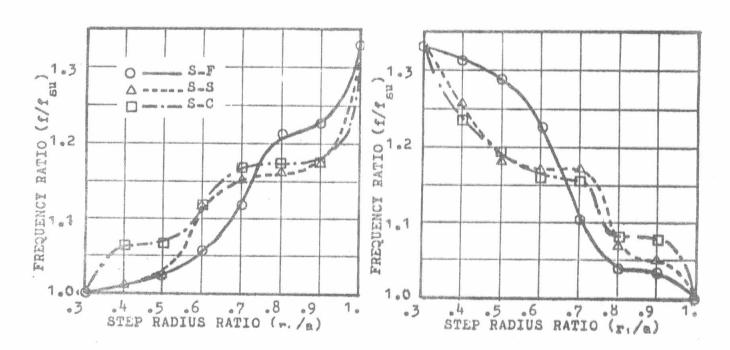
(a) (b)
Fig.4 Natural frequencies of S-F, S-S and S-C stepped annular plates at the first symmetric mode 00.

a- plates with central region raised.b- plates with central region unraised.





(a)
(b)
Fig.5 Natural frequencies of C-F, C-S and C-C stepped annular plates at the second symmetric mode O1.



(a)
Fig.6 Natural frequencies of S-F, S-S and S-C stepped annular plates at the second symmetric mode O1.

a- plates with central region raised.b- plates with central region unraised.



In the case of central region raised, when the step radius r, equal to the inner radius b, the plate will be with uniform thickness of 0.75 mm while when the step radius equal to the outer radius a, the plate will be with uniform thickness of 1.0 mm. Consequently the vice-versa will be in the case of central region unraised. Table 1 shows the frequency results for these annular plates of uniform thicknesses and its comparation with those obtained by Raju [4]. The results were convenient and useful for the application of this process in a large scale.

Table 1. Natural Frequencies of Uniform Thickness Annular Plates.

Type of Mode Boundaries mn	Natural Frequencies (Hz) by			
	Raju [4]		Given Method	
	h=0.75 mm	h=1.00 mm	h=0.75 mm	h=1.00 mm
$C = F \begin{cases} 00 \\ 01 \\ 0 \\ 0 \\ 01 \end{cases}$ $C = S \begin{cases} 00 \\ 01 \\ 00 \\ 01 \end{cases}$ $S = F \begin{cases} 00 \\ 01 \\ 01 \\ 01 \\ 01 \end{cases}$ $S = G \begin{cases} 00 \\ 01 \\ 01 \\ 01 \end{cases}$	27.45 124.50 81.07 250.50 109.26 301.05 11.17 89.10 50.77 197.02 71.92 240.82	36.60 166.00 108.10 334.00 145.69 401.40 14.90 118.80 67.80 262.70 95.90 321.10	27.00 122.00 80.00 246.00 107.20 296.00 11.00 87.00 50.00 194.00 71.00 237.50	36.00 163.00 107.00 329.00 144.12 395.00 14.60 117.00 67.00 260.00 95.00

CONCLUSIONS

The analysis of the axisymmetric vibrations of stepped annular plates are presented for plates with different combinations of boundary conditions. It is seen that in the case of plates with central region raised, the value of the natural frequency usually increases as the step radius increases. In the case of plates with central region unraised the vice-versa will happened except that there are some of the adjacent step radii at which the corresponding frequencies can be considered equal. Also the given charts show the effect of the boundary conditions on the results from the design point of view.

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NOMENCLATURE

B Outer radius of the annular plate E Modulus of elasticity. b Inner radius of the annular plate. f Natural frequency. h Plate thickness. Modified Bessel functions of the first kind of order 0, and 1. J_0 , JBessel functions of the first kind of order 0 and 1. Ko,K, Modified Bessel functions of the second kind of order O and 1. m No. of nodal diameters. M Radial bending moment. n No. of nodal circles. Q Transverse shear force. r Plate radius. W Plate deflection. Yo, Y, Modified Bessel functions of the second kind of order O and 1. Eigenvalue of the frequency equation. B P Density of the plate. v Poisson's ratio. θ Angular displacement. Natural frequency.